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A Shape Memory Alloy-Based Morphing Axial Fan Blade— Part I: Blade Structure Design and Functional Characterization

The possibility to realize adaptive structures is of great interest in turbomachinery design, owing to the benefits related to enhanced performance and efficiency. To accomplish this, a challenging approach is the employment of shape memory alloys (SMAs), which can recover seemingly permanent strains by solid phase transformations whereby the so-called shape memory effect (SME) takes place. This paper presents the development of a heavy-duty automotive cooling axial fan with morphing blades activated by SMA strips that works as actuator elements in the polymeric blade structure. Concerning the fan performance, this new concept differs from a conventional viscous fan clutch solution especially during the nonstationary operating condition. The blade design was performed in order to achieve the thermal activation of the strips by means of air stream flow. Two polymeric matrices were chosen to be tested in conjunction with a commercially available NiTi binary alloy, whose phase transformation temperatures (TTRs) were experimentally evaluated by imposing the actual operating thermal gradient. The SMA strips were then thermomechanically treated to memorize a bent shape and embedded in the polymeric blade. In a specifically designed wind tunnel, the different polymeric matrices equipped with the SMA strips were tested to assess the fluid temperature and surface pattern behavior of the blade. Upon heating, they tend to recover the memorized shape and the blade is forced to bend, leading to a camber variation and a trailing edge displacement. The recovery behavior of each composite structure (polymeric matrix with the SMA strips) was evaluated through digital image analysis techniques. The differences between the blade shape at the initial condition and at the maximum bending deformation were considered. According to these results, the best coupling of SMA strips and polymeric structure is assessed and its timewise behavior is compared to the traditional timewise behavior of a viscous fan clutch. [DOI: 10.1115/1.4031272]

23 Introduction

24 Actuators are devices which perform a task, such as moving an 25 object, either on demand or in response to certain changes in their 26 environment (temperature, pressure, etc.). In a modern car, more 27 than one hundred actuators are used to control engine, transmis-28 sion and suspension performance, to improve safety and reliability 29 and enhance driver comfort [1]. Most of these actuators today are 30 electric motors and solenoids. For this reason, the control systems 31 account for the majority of the weight and volume of vehicle com-32 ponents and in some cases, they are too bulky, expensive, and not 33 sufficiently robust for the intended application.

Renewed interest in automotive control systems is especially due in order to limit fuel consumption and exhaust emissions. More than half of the energy in vehicles is lost as heat to the different cooling systems (engine, driver, and passenger compartment space and auxiliary devices) and exhaust gas. Reducing the amount of energy lost in vehicle cooling systems will enhance the efficiency of the vehicles [2].

Technological progress has ensured the obtainment of high
 efficiency levels as a result of the real-time performance evalua tion. The entire control system is therefore optimized, even in
 nonstationary operating conditions. The integration of smart mate rials in actuation systems represents an excellent technological

opportunity for the development of simple, very compact, and reliable actuator devices. These structures are thus transformed from static to dynamic or, in some cases, adaptive as they can react directly to environmental stimuli. Smart materials can simplify products, add new functions, upgrade performance, improve reliability, and reduce component cost, mechanical complexity, and size. 52

53 The present paper focuses on an innovative passive control 54 system for the performance optimization of heavy-duty automo-55 tive cooling axial fan. This challenging approach enables an optimal response to possible changes in turbomachinery operating 56 57 conditions, avoiding any external actions on the shaft rotational velocity. The system is regulated by a sensorless control taking 58 advantage of SMAs elements, whose phase transformation ena-59 bles the production of favorable aerodynamic blade shape 60 61 changes, according to the air flow temperature. Unlike conventional actuation and control systems employed in cooling fans 62 (i.e., on/off clutch and air sensing modulated viscous clutches), 63 the SMA actuation allows to control and adjust the fan perform-64 ance to the engine thermal requests. This fan's setup can allow an 65 almost continuous operation of the turbomachinery in the 66 67 maximum efficiency point and not only for a discrete number of operating points. Given that the thermally activated phase trans-68 formation does not occur instantly, the SMA enables the develop-69 70 ment of heavy-duty machines which continuously modulate their working point, following the requested operating change seam-71 72 lessly. As a result, the passive SMA-controlled system allows to: 73 (i) eliminate the active controls (electric motor, thermostats and 74 valves, and temperature probe), (ii) adapt the cooling fan to the

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engine thermal requests, (iii) change the performance independ-76 ently from the engine rotational velocity, (iv) reduce the coolant 77 warmup time during cold engine starts, and (v) eliminate the 78 heat-soak issue. Nevertheless, the nonlinear and hysteretic ther-79 momechanical responses of the SMA actuators prevent, to date, 80 the precise positional control and hence compensation techniques 81 involving proportional and derivative control are often employed. 82 This two-part work addresses the reliability of the SMA ther-83 moresponsive actuators for a heavy-duty automotive cooling axial 84 fan. The selection and characterization of the SMA and polymeric 85 materials presented in this first part were aimed at developing the 86 blade structure design. To assess the morphing capability of the 87 composite structure, experimental tests were carried out in a 88 purpose-built wind tunnel. This experimental characterization 89 was intended to provide an evaluation of the reliability of an 90 SMA-based morphing axial fan blade and a comparison with a

91 traditional viscous fan clutch.

92 Literature Survey

Research on possible ways to control the automotive cooling
fan has been proposed since the 1970s [3]. Cooling fans can be
actuated and controlled by (i) a friction-type fan clutch (commonly named on/off clutch) and (ii) an air sensing modulated
viscous clutch.

The on/off clutches were most widespread in the past especially in North America for heavy-duty vehicles. On/off clutches operate by design with the fan either at an idle rotational velocity (typically ± 200 rpm) or at a fully engaged rotational velocity (± 2000 rpm). These types of clutches penalized the power efficiency in providing engine cooling [4].

104 The modulated viscous clutches have a continuous velocity 105 control that provides the proper cooling air flow rate proportional 106 to engine cooling load by means of a bimetal element [5]. The 107 bimetal element senses the air temperature approaching the drive 108 from the radiator opening and closing a valve spring to either store 109 the fluid (low fan velocity) or allow full flow to the working area, 110 providing maximum fan velocity. The bimetal element expands 111 and contracts in proportion to the variation in temperature. The 112 two basic benefits derived from the utilization of fan clutches are 113 reduced horsepower drainage to the fan and a reduction in the 114 average noise output of the fan.

As reported in Ref. [4], the average power consumption of an
on/off friction clutch is higher than that of a modulating fan clutch
providing the same average cooling rate. A modulating fan clutch
theoretically consumes only 22% of the power of an on/off clutch,
providing equivalent average cooling during a duty cycle, which
requires on/off fan engagement 20% of the time.

121 Even if the modulated viscous clutches allow the control of the 122 fan velocities, the proper management of the engine cooling rate 123 and, in particular, of the engine temperature must be realized by 124 linking the coolant engine temperature with the cooling capacity 125 of the fan. The clutches join the engine shaft with the cooling fan 126 which does not follow the engine thermal requests strictly since it 127 is closely connected with the engine rotational velocity, without 128 being affected by the airflow temperature. Also, engine encapsula-129 tion increases the retarders that lead to unacceptable time response 130 lag from indirect sensing in new generation vehicles [4].

In fact, in some cases the electric driven fans have replaced the
clutch-driven fan and the rotational fan velocity is controlled by
some sensors positioned in the cooling circuit [6]. This approach
improves the engine thermal management but, on the other hand,
by using many sensors and devices makes the control system
more complicated.

Driven by the need to maximize overall system performance,
engineers and designers have sought to increase the multifunctionality of several design components. As a result, the so-called
stimulus-responsive materials have increasingly captured worldwide interest. Among these, the SMAs represent a challenging
solution for a wide range of engineering applications [7]. The

SMAs are a class of metallic materials with the ability to recover 143 seemingly permanent strains, as a result of a temperature and/or 144 stress induced solid phase transformation. The reversible crystal- 145 lographic phase transformation of the SMAs has been widely used 146 in solving engineering problems where the actuation purposes 147 have been fulfilled by the SME. When a constraint condition is 148 applied to an SMA material, the phase transformations are associated with a significant reversible deformation capability, which 150 leads to the generation of considerable stresses. 151

152 As a result, SMA elements are attractive especially for the development of aerodynamic applications where this actuation 153 solution prevents the introduction of flow-disturbing control elements [8]. The thermal and mechanical properties of SMA allow 155 new design solutions for actuators, structural connectors, vibration dampers, sealers, release or deployment mechanisms, inflatable 157 structures and manipulators [8,9]. In these devices, SMAs in the 158 form of wires or strips are usually embedded in thermoplastic and 159 thermosetting polymer matrices to work as active elements. 160 SMA-based actuators provide high force to weight ratio, long 161 fatigue life, and high corrosion resistance; hence, these materials 162 have been employed in many applications such as active helicop-163 ter rotor blades, adaptive airfoils, and deployment of control surfa-164 ces and flaps [10-12]. Several studies [13-16] deal with the use of 165 SMAs as linear actuators, to realize reconfigurable airfoils which 166 enable an increase in the efficiency of the wing in flight at several 167 different flow regimes. Recently, Sofla et al. [17] have proposed 168 the design of a shape morphing wing for small aircraft which 170 takes advantage of an antagonistic SMA-actuated flexural structural form that enables the changing of the wing profile by bend-171 172 ing and twisting, thus improving the aerodynamic performance. More recently, the Boeing Company has developed an active 173 aerodynamic device, known as variable geometry chevron, able to 174 reduce noise during takeoff and able to increase cruise efficiency 175 176 [18]. The SMA morphing capability in conjunction with the simplicity and compactness of active deformable structures provide 177 178 substantial performance benefits. Compared to other types of standard actuation, an SMA control system allows the design of 179 devices with reduced complexity, higher overall reliability, 180 easier serviceability, cheaper implementation, and more compact 181 182 arrangement in conjunction with improved lightness [7].

AO2

The notion of smart advanced blades, which can control 183 themselves and reduce (or eliminate) the need for an active con-184 185 trol system, is a highly attractive solution in blade technology. Current systems based on morphing or adaptive blades, beyond 186 aerospace applications, are used in wind turbine applications, 187 where fast actuation without complicated mechanical systems and 188 189 large energy-to-weight ratios are required. In such cases, it is important to save in weight and complexity in the rotor design and 190 its auxiliary mechanisms while also reducing the costs of energy 191 192 generation [19,20]

Unfortunately, specific studies on fan performance modification 193 194 using the SMA devices are scarce or even not available in literature. In addition, the variation of the blade shape based on the 195 196 strain provided by the SMAs is not widely investigated. The 197 objective of this study is to assess the capabilities of the SMA-based morphing blade, which is inherently deformable and 198 199 adaptable, to change its shape and consequently the local flow field to enhance overall fan performance. The SMA phase trans- 200 formation leads to camber variation and trailing edge displace-201 202 ment, which allows the modulation of the blade shape.

This work focuses on the development of an SMA activated 203 blade used in a heavy-duty automotive cooling axial fan. The 204 thermal activation of SMA elements could be used to control the 205 cooling fan performance in order to adapt the coolant fluxes with 206 engine thermal requests. Actual thermal requests and fan performance represent the starting point of the blade design. 208

This first part focuses on the blade development and discusses 209 the capability of the SMA elements to modify the blade shape. 210 This paper develops according to the following points: (i) defini- 211 tion of the thermal requirement in heavy-duty applications, (ii) 212

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213 selection and characterization of the SMA elements, (iii) selection 214 of the polymeric structure (iv) characterization of the blade struc-

of the polymeric structure, (iv) characterization of the blade structure (polymeric matrix with the SMA strips) through experimental

tests developed in a wind tunnel with an evaluation of the reliabil-

217 ity of the SMA blade, and (v) comparison between the SMA

activation and the viscous clutches in terms of reaction time and

²¹⁹ time-lag with respect to thermal input.

220 Working Conditions

AO3

221 The primary focus of this work is to provide the control of a 222 heavy-duty automotive cooling axial fan. In this application, the 223 rotational velocity of the fan is almost constant, but the engine 224 load changes according to the working condition of the operating 225 machine. For these reasons, the thermal energy that must be 226 removed from the engine changes during operation and the cool-227 ing fan must be designed in order to remove the thermal energy at 228 the engine load peak [21]. A typical temperature variation for a 229 heavy-duty engine is reported in Fig. 1 that is related to the tem-230 perature measurements reported in Ref. [22] for a heavy-duty 231 diesel engine TDC 6V2015 (6 cylinder, engine displacement of 232 121) during warmup.

233 The temperature trend reported in Fig. 1 refers to the cooling 234 water temperature and represents only the typical temperature var-235 iation that occurs in a typical cooling circuit. It can be noticed that 236 the maximum temperature does not exceed 95 °C and the cooling 237 water reaches a stable temperature at about 420 s from the engine 238 start. The steep ramp and the following sinusoidal temperature 239 variation are due to the action of the thermostat valve that controls 240 the water flow rate through the cooling circuit. For the aim of this 241 study, the temperature gradient that characterizes the engine warmup is related only to the temperature difference between the 242 243 start and end warmup temperature compared to the warmup time. 244 From the temperature and time data reported in Fig. 1, it can be 245 seen that the temperature gradient, obtained as the angular coefficient of a liner interpolation between the first point and the final 246 247 point of the engine warmup, is equal to about 9 °C/min.

248 Blade Structure Design

249 The morphing blade design was performed with the aim of real-250 izing a functional composite structure which allows the control 251 the working condition parameters of the axial fan, taking advant-252 age of the SME. To accomplish this, the SMA strips were used as 253 actuator elements, embedded in the polymeric blade structure. 254 The intent to use the embedded SMA elements refers to realize a 255 fan in which the blade shape changes continuously as a function 256 of an external stimulus such as the airflow temperature. In order to 257 study the capability of the morphing blade to adapt its shape 258 according to the fan's operation temperature, the SMA thermal 259 activation was achieved by a hot/cold air stream flow. The

100 7 (°C) 80 end of warm up process 60 40 20 120 240 360 480 t (s) 600 Fig. 1 Temperature trend during the engine warmup [22]

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embedded strips were put in contact with the fluid flow by means 260 of several slots. 261

SMA Strip Selection. The choice of the SMA compound is 262 one of the most important steps in the fan's design. The best SMA 263 compound must have the closest TTR to those encountered by the 264 fan during the operation. For this reason, the commercially avail- 265 able NiTi SMA (Memry Metalle Company) of nominal composi-266 tion $Ni_{50,2}Ti_{49,8}$, as a 1.5-mm thick plate was chosen. Starting 267 from the supplied foil, the strips were cut by means of electroerosion machining in order to minimize microstructural alterations 269 resulting from thermomechanical stresses induced by cutting 270 processes. 271

As mentioned above, the SMA materials offer advantages in 272 terms of high reversible strains, high damping capacity, wide 273 reversible changes of mechanical and physical properties, and the 274 ability to generate extremely high recovery stresses (upto 275 800 MPa). Among the different SMA compositions, the near 276 equiatomic NiTi alloys are by far the most widely used shape 277 memory materials for engineering applications. The SME is the 278 property of the material to recover mechanically induced strains 279 (upto 10%) when it is deformed in the low temperature phase $\frac{280}{2}$ 281 (martensite) and subsequently heated to the high temperature phase (austenite). This thermal change forces the return to the aus-282 283 tenite and brings the SMA to its original macroscopic shape. In the martensitic phase, SMA can be easily deformed according to 284 the lattice arrangement. In the stress-free condition, the transfor-285 mation from austenite to martensite occurs during cooling: it 286 287 begins at the martensitic start temperature M_s and ends at the martensitic finish temperature Mf. Conversely, the reverse transforma-288 289 tion, from martensite to austenite, occurs upon heating: this begins at the austenitic start temperature A_s and ends at the austen-290 itic finish temperature $A_{\rm f}$. These four temperatures are known as 291 292 TTRs

According to the ASTM F2004 standard, differential scanning 293 294 calorimetry (DSC) tests, by means of a TA Instruments DSC Q2000, were carried out on a small fraction of the strips. A por-295 tion of the untreated material was chosen to be characterized in 296 297 order to obtain useful quantitative information for the shape memory treatments described below. For the DSC measurements, a 298 constant heating/cooling rate of 10°C/min was set. This thermal 299 gradient is comparable with the actual thermal gradient in the 300 301 automotive cooling circuits (see Fig. 1) and allows the material characterization in the actual operating condition. A complete 302 thermal cycle (heating/cooling) of the obtained phase transforma-303 tion is given in Fig. 2. The characteristic TTRs were extrapolated 304 from the DSC plot through the tangential line method (intersec-305 tions of a baseline and the tangents to each peak) [23]. The experi-306 307 mental values are summarized in Table 1. As mentioned, TTRs 308 are fundamental for the development of the thermomechanical treatment used to memorize the shape, described as follows. 309

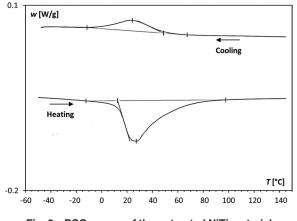


Fig. 2 DSC curves of the untreated NiTi material

MONTH 2015, Vol. 00 / 000000-3

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Table 1 TTRs of SMA material in the untreated condition

10 °C
57 °C
48 °C
−4 °C

310 In Fig. 3, a representative scheme of the treatment is depicted. 311 In order to delete any residual stress of previous deformation history, the samples were first placed in a tube furnace and 312 313 annealed at 700 °C for 20 min followed by controlled cooling to 314 room temperature. The development of the best thermomechani-315 cal treatment for memorizing the defined bent shape, which allows 316 the bending of the blade, was experimentally carried out. Temper-317 ature and time parameters were chosen according to the results of 318 a previous study, where it was experimentally found that heating 319 the material at 450 °C for 25 min represents the best shape mem-320 ory setting, allowing 92% of shape recovery [24]. To memorize 321 the bent shape, the strips were subjected to a double thermome-322 chanical treatment. After the annealing, they were first strained in 323 the martensitic state by immersion in a propylene glycol bath 324 cooled to -15 °C. The strips were strained and wound on a cylin-325 drical jig to reach a circular shape. This setup was then placed 326 into a tube furnace in constrained conditions, in order to avoid the 327 shape recovery on heating. To memorize this first shape, the mate-328 rial was heated at 450 °C for 25 min and subsequently quenched in 329 the propylene glycol bath cooled to -15 °C. After this treatment, 330 the strips were again strained in the martensitic state (performed 331 during the immersion in the propylene glycol bath cooled to 332 -15 °C) applying opposite bending couples acting at the ends, 333 and locked into a specifically designed arc clamp. To memorize 334 this bent shape, they were again thermally treated, at the 335 aforementioned temperature and time conditions. Finally, the 336 heat-treated NiTi strips were strained to a flat shape, applying a 337 uniform bending load, to be embedded in the blade structure.

338 Polymeric Structure Selection. The blade shape is a typical 339 forward-swept blade used in partially shrouded cooling fans for 340 automotive applications. The sketches of the fan setup and the 341 blade shape are reported in Fig. 4. The polymeric blade design 342 was performed with the aim of being produced by means of injec-343 tion molding. Since SMA strips induce the necessary bending 344 force on the polymeric blade structure when thermally activated 345 by air stream flow, they were located into purpose-built slots in 346 direct contact with the stream flow. The two slots gather from the 347 blade polymeric structure does not modify the blade surface and 348 does not influence the aerodynamic performance of the airfoils. 349 According to the most widely used polymeric matrices in blade 350 fan production, two polymeric mixtures of Nylon PA 6.6, glass 351 fibers and elastomer, were chosen to be tested: Compound A 352 had a lower amount of glass fiber reinforcement compared to 353 Compound B.

The key issue was to realize a blade capable of withstanding the prescribed loads and also able to change its shape. The

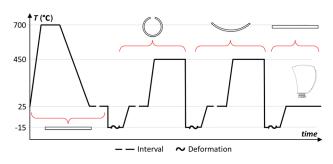


Fig. 3 Representative scheme of the SMA thermomechanical treatment

000000-4 / Vol. 00, MONTH 2015

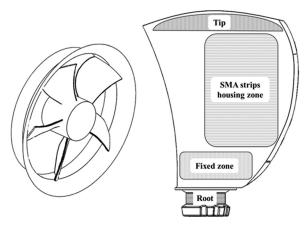


Fig. 4 Fan setup and blade shape sketches

structure was designed in order to be sufficiently compliant and 356 flexible to support the large deflections induced by the strips and 357 to allow the shape recovery, but also stiff enough to withstand the 358 aerodynamic loads. 359

In Fig. 4, the blade sketch, with the essential region, is reported. 360 As can be seen, the embedded SMA strips were located in the 361 range of about 50–85% of the blade span and next to the trailing 362 edge, in order to achieve the desired deflection upon activation. 363 Conversely, the region near the root is depicted as fixed since no 364 actuation elements were placed. The position of the SMA strips 365 realizes a camber curvature variation of the airfoil: blade shape 366 modification allows the variation of the fan performance analyzed 367 by computational fluid dynamics (CFD) numerical simulation, 368 proposed in the second part of this work [25]. 369

AO4

370

Experimental Apparatus

In order to test the two selected polymeric blade structures 371 under the SMA loading, specific experimental tests were 372 conducted in a purpose-built wind tunnel, named single blade test 373 facility (SBTF), where it was possible to measure the air tempera-4 ture and velocity, the surface blade temperature and to detect the blade shape changes. The blade was positioned according to the flow direction that represents the relative flow velocity in the real operating conditions. 378

SBTF Description. As depicted in Fig. 5, the SBTF was com- 379 posed of (i) a convergent device, (ii) a polyvinyl chloride pipe 380 with a diameter of 250 mm and length of 3000 mm, (iii) a flow 381 straightener, (iv) a polymethyl methacrylate transparent measure- 382 ments section with a square section of $250 \text{ mm} \times 250 \text{ mm}$ and 383 1000 mm in length, and (v) exhaust pipe with a diameter of 384

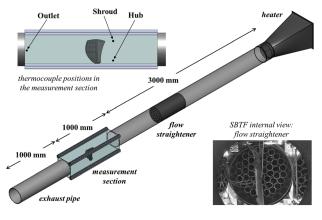


Fig. 5 SBTF functional scheme

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 385 250 mm and 1000 mm in length. The wind tunnel was driven by an axial fan with a nominal 1500 m³/h flow rate.

387 A 22 kW-electric heater allowed the obtainment of an air flow 388 stream characterized by a highly reproducible timewise thermal 389 gradient, which can reach values of up to about 12 °C/min in heat-390 ing mode and up to about 6°C/min in cooling mode. These tem-391 perature gradients are consistent with (i) the operating conditions 392 of the fan when used in its normal duty (see Fig. 1) and (ii) the 393 constant heating/cooling rate used for the SMA material charac-394 terization (see previous paragraph).

395 In order to evaluate the thermofluid dynamic conditions, a hot 396 wire anemometer (for cold conditions), a pitot static tube (for hot 397 conditions), and several calibrated thermocouples were installed 398 in the SBTF. A pitot static tube (Velocical Plus, TSI) and hot 300 wire anemometer (Velocical Plus, 964 TSI) were placed at the 400 inlet of the transparent measurements section. The accuracy of the 401 velocity measurements is equal to 3% of the reading. The velocity 402 field along the blade span was continuously measured to verify 403 the uniformity of the air velocity during the test provided by the 404 flow straightener position in the second half section between the 405 heater and the measurement section.

406 Mineral insulated thermocouples type K were placed in corre-407 spondence to the heater, in the neighborhood of the blade (at the 408 shroud and hub positions) and at the outlet section as can be seen 409 in Fig. 5. Several welded tip thermocouples type K were also 410 placed on the blade surface and on the SMA strips to acquire the 411 temperature variation. The control was performed by a National 412 Instrument NI 9213 thermocouple measurement device and a 413 LABVIEW software acquisition. The thermocouples are individually 414 calibrated in a thermostatic bath against a reference temperature 415 sensor and a first-order linear calibration curve is obtained in the 416 range of 288-480 K. The accuracy of these sensors is estimated as 417 equal to ± 0.5 K.

418 Thanks to the transparency of the measurements section, the 419 modification of the blade shape was continuously evaluated by 420 means of digital image analysis techniques. Three digital cameras 421 were aligned in correspondence to the blade tip, suction side, 422 and pressure side, respectively. The temperature trends were 423 synchronized with the video acquisition (1024×768) pixels in 424 order to control the overall shape evolution related to the tempera-425 ture trend. The resolution of the video acquisition allows a spatial 426 resolution up to 0.16 mm that is considered suitable for evaluating 427 the blade shape modification during the activation tests.

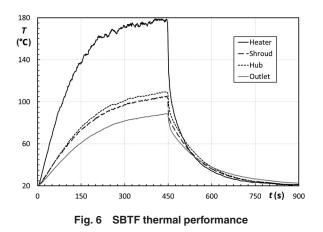
428 Thermal Cycle. The SMA thermal activation and the resulting 429 blade deflection were achieved by (i) a heating ramp and (ii) a 430 cooling ramp, described as follows. Starting from the room tem-431 perature, the blade was first heated by a hot air stream flow which 432 caused the activation of the SMA strips and the blade deflection. 433 The blade reached the maximum deflection as the fluid flow 434 reached the maximum temperature. Subsequently, the blade was 435 cooled down to room temperature in order to achieve the 436 martensitic phase transformation.

437 Figure 6 shows the experimental temperature trends as a 438 function of the time in correspondence to the sections illustrated 439 in the sketch of Fig. 5. As can be seen, it was possible to achieve 440 uniform thermal conditions of the air flow stream on the blade 441 during the execution time, both in the heating ramp (maximum 442 hub-to-shroud temperature difference of about 1.5 °C) and in cool-443 ing ramp (maximum hub-to-shroud temperature difference of about 0.4 °C). 444

445 Blade Activation

As mentioned above, to achieve the phase transformation from martensite to austenite, the heating ramp has to follow a prescribed temperature gradient and has to reach the designed temperature peak. In Fig. 7, the temperature trends, for the polymeric structure and SMA strips, measured by using the welded tip thermocouples are reported. Given that uniform thermal conditions of





the air flow stream and the high-reproducibility of the temperature 452 trend on the blade were achieved by the realized SBTF, the 453 depicted trends are representative for the two considered poly- 454 meric matrices. Temperature trends reported in Fig. 7, for poly- 455 meric structure and SMA strips, were obtained with Compound A. 456 As can be seen from Fig. 7, the temperature gradient in both the 457 polymeric matrix structure and the SMA strips are quite similar 458 and the blade shows an almost uniform surface temperature pat- 459 tern. The temperature-time trends measured on the blade surface 460 are comparable with those measured in the SBTF (see Fig. 6) 461 revealing that the temperature gradient realized by the SBTF is 462 suitable for the aim of the present study. The average value of the 463 air flow temperature measured in the hub and shroud position (see 464 Fig. 5) is reported in Fig. 7 by using grey diamonds. During the 465 heating ramp, when the temperature of all the SMA strips reached 466 80°C they tended to recover the memorized bent shape and the 467 blade structure was forced to bend. At the peak temperature, the 468 SMA strips induced the maximum blade deflection. To quantita-469 tively evaluate the blade deformation on thermal activation, digi- 470 tal image analyses were performed. The final deformed shape of 471 the blade is the results of the combination of the load provided by 472 the SMA strips and the stiffness (or strength) provided by the 473 polymeric matrix. The blade shape change in terms of mean line 474 deflection, develop on each blade-to-blade plane as a function of 475 the blade span location. As a result, the centrifugal force that 476 works in the actual blade's operation does not influence the blade 477 shape modification. The SMA strips determine the airfoil deflec- 478 tion along the chordwise direction without being affected by the 479 centrifugal force that works along the blade height. An aerody- 480 namic evaluation of the blade deformation is reported in the sec- 481 ond part of this work [25]. The preliminary design of this 482 composite structure takes into consideration the activation 483

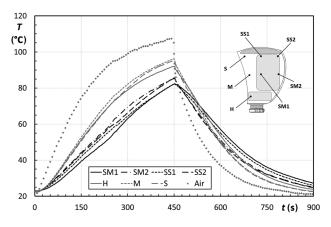


Fig. 7 Temperature trends for Compound A: polymeric structure (H, M, and S) and SMA strips (SM and SS)

MONTH 2015, Vol. 00 / 00000-5

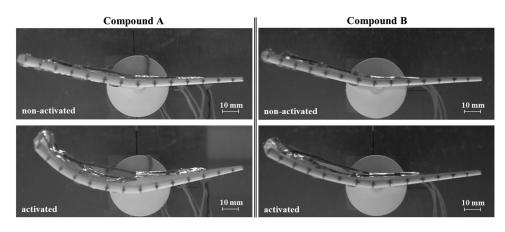


Fig. 8 Digital captures from recorded video at blade tip view

484 temperature in the stress-free condition (as mentioned in previous 485 paragraph) and only by the wind-tunnel test it is possible to evalu-486 ate a posteriori the aerodynamical changes in the blade shape. In 487 Fig. 8, the comparison between the two blade shapes, both for 488 Compound A and Compound B, is reported: (i) initial (nonacti-489 vated) condition and (ii) maximum deflection (activated) captured 490 from the blade tip view. From the comparison of the two activated 491 blades, it is evident that Compound A reaches the highest strain, 492 according to its composition. Conversely, the greater amount of 493 glass fiber in Compound B provides more stiffness which cause a 494 smaller deformation compared to Compound A. The deformation 495 in both cases is localized in the trailing edge zone.

496 Taking into account the behavior of Compound A, when the 497 temperature reaches its peak (activated condition), it can be seen 498 that the greatest deformation is localized at the trailing edge where 499 the deflection of the SMA strips imposes the maximum strain on 500 the polymeric structure. Also, on the first part of the blade chord, 501 the blade shape changes according to the memorized bent shape. 502 This modification determines a variation of incidence angle 503 during the operation, and consequently, a variation of the fan 504 performance.

505 To study the blade deformation achieved by the shape recovery 506 of SMA strips, a quantitative analysis of the deformations was 507 performed. A CAD software reconstruction of the shape at the 508 nonactivated and activated conditions is proposed in Fig. 9. Thus, 509 the evaluation refers to the airfoil shape at the blade tip. Note that 510 the airfoil deformation was measured as the maximum distance 511 (maximum camber) from the chord length (leading edge to trail-512 ing edge line). Compound A (21.4 mm) showed a higher airfoil 513 deformation than Compound B (14.2 mm), equal to about 40%. 514 The higher percentage of glass fiber in Compound B restricts the 515 recovery shape capability of the SMA strips. According to the 516 experimental results, Compound A accomplished the largest 517 deflection induced by activation of the SMA strips and therefore 518 this polymeric mixture was chosen for the realization of the blade 519 structure.

520 Subsequent to the heating, the cooling ramp to room tempera-521 ture was achieved by the supplied air provided by the fan. This 522 allowed the transformation from austenite phase to martensite 523 phase and the subsequent return to the initial condition

Compound A 0.3 non-activated 21.4 activated [mm] Compound B 0.3 non-activated 14.2 activated [mm]

Fig. 9 Blade tip superimpositions for activated and nonactivated conditions

(nonactivated). Since the polymeric structure could be affected by 524 a viscouselastic behavior, some sensitivity tests were conducted 525 by the authors. Different cooling gradient temperatures were studied (up to $12 \,^{\circ}$ C/min) and no viscouselastic effects were noticed in 527 the blade structure. In the real operating conditions, the cooling 528 ramp refers to (i) the unloading condition of the engine and (ii) 529 the thermal gradient that still exist in the engine after its stop. The 530 cooling ramp is usually less steep than the heating ramp and the 531 results obtained in the cooling mode guarantee the blade structure 532 functionality. 533

Time-Lag Comparison

In a traditional cooling system, the energy optimization could 535 be achieved by controlling the engine temperature and reducing 536 the cooling fan run time by using a fan clutch. The operation of 537 the fan is controlled by means of some sensors placed in the cool-538 ing circuit for measuring the air temperature and thus tuning the 539 rotational fan velocity. Nevertheless, the increasingly utilization 540 of a great number of sensor devices and the resulting raised com- 541 plexity make, in some cases, the control system too bulky, expen- 542 sive and not sufficiently robust for the intended application. 543 During the actual fan cooling operation, the air temperature 544 changes according to the engine load and/or the effect of the ram 545 546 air and, at the same time, the fan rotational velocity could be 547 changed due to the engine operation load requirement.

Conversely, in the present study the performance modification 548 during operation is completely obtained by the blade shape 549 modification provided by the SMA elements without sensors and 550 control systems. 551

To study the possibility of employing SMA strips as actuator 552 elements, a comparison with common viscous clutches behavior 553 is proposed. In Fig. 10, the timewise evolution of air temperature, 554 airfoil camber at the blade tip, and rotational fan velocity are 555 depicted. To highlight the differences between the two actuating 556 solutions, the comparison between the timewise camber variation 557 and the timewise rotational velocity variation is performed with 558 the same timewise temperature variation. The thermal variation 559 trend starts from 0s and ends at 440s, and it is inline with the 560 heating ramp reported above (see Fig. 6) and similar to a common 561 562 engine coolant temperature variation during the warmup condition. 563

The comparison is related to (i) the evaluation of the time lag 564 that represents the actuator awaiting time after the thermal input 565 (green lines in Fig. 10) and (ii) the evaluation of the time range 566 transient actuator response (time ranges between green lines and 567 red lines). The considered viscous clutch is an on/off viscous fan 568 clutch but similar results can be obtained to considering a modu-569 lated viscous fan clutch. As reported by Everett [26], modulated 570 viscous clutch allows the reduction of the power consumption but 571 the time range transient actuators response is even so very limited 572

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000000-6 / Vol. 00, MONTH 2015

Transactions of the ASME

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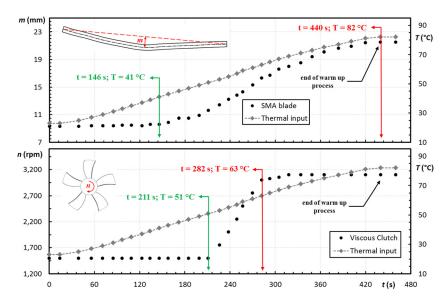


Fig. 10 Timewise evolution of air temperature, airfoil camber at the blade tip and rotational fan velocity

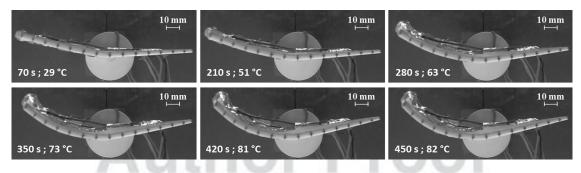


Fig. 11 Blade shape evolution during the heating ramp at the tip view

573 (with a thermal input range of $60 \degree \text{C}-71 \degree \text{C}$ (140 $\degree \text{F}-160 \degree \text{F}$) the

fan speeds moves from 1000 rpm to 2000 rpm).

575 As can be seen from Fig. 10, the SMA time lag is shorter than 576 the viscous clutch, 146 s and 211 s, respectively. Then the SMA 577 actuator requires a lower temperature to start the reaction that 578 occurs 10 °C earlier than the viscous clutch. Regarding the time 579 range transient actuator response, it is evident that the viscous 580 clutch solution provided a fast change of the fan rotational veloc-581 ity, as can be seen from the steep ramp in Fig. 10. This step 582 change velocity variation does not follow the thermal input and 583 the fan rotational velocity reaches the target velocity independ-584 ently from the thermal input variation after 12 s. The fan rotational 585 velocity reaches a value lower than the rotational velocity of the 586 driver shaft. In this specific case, the rotational velocity of the 587 driver shaft is equal to 3300 rpm compared to the target rotational 588 velocity equal to 3000 rpm.

589 Conversely, the smooth change camber variation follows the 590 thermal input thanks to thermal controlled SMA strips actuation. 591 The transient SMA actuator response ends at 440 s which corre-592 sponds to the thermal input end. This experimental result confirms 593 the capability of SMA materials to cover the lower power actua-594 tors in the automotive field. The time range actuator response indi-595 cates that the SMA strips provide a lower frequency control that 596 fits well with engine thermal requirement. The SMA strips blade actuator could improve the match between the fan performance 597 598 and engine cooling thermal request and thus enhance the engine 599 thermal management.

In order to emphasize the progressive blade deflection provided by SMA strips activation on heating, the blade tip views are showed in Fig. 11. Each blade tip capture reports the time instant 602 and the average value of the SMA strip surface temperature. The 603 time steps capture highlights the smooth camber variation accordfing to the temperature increment. Due to this progressive blade 605 deflection, the consequent fluid dynamic phenomena evolve durfing the heating ramp and, for this reason, in the second part of this 607 work [25], the authors have reported an extensive analysis of the 608 blade shape. The engine coolant temperature control through the 609 use of a fan with the SMA activated blades allows the adjustment 610 as a function of the thermal request. This adjustment results from 611 the selection of (i) the memorized shape of the SMA strips and (ii) 612 the polymeric mixture used in the polymeric matrix. 613

Conclusions

614

In this paper, the development of a morphing axial fan blade 615 via SMA strips activation has been proposed. Commercially avail- 616 able NiTi strips were characterized and two polymeric blade 617 structures were tested. 618

The thermal characterization of the SMA material allows the 619 study of the thermomechanical treatment in order to reach the 620 suitable bent shape. The thermomechanical treatment parameters 621 (temperature, time, and dedicated clamp) have been experimenfally tuned to maximize the SME in the NiTi strips. The SMA 623 strips were subsequently embedded in two polymeric structures 624 with different amounts of glass fiber reinforcement to study their different bending ability. 626

Several experiments for each compound were performed by 627 using a purpose-built wind tunnel and the blade modifications 628

Journal of Engineering for Gas Turbines and Power

MONTH 2015, Vol. 00 / 00000-7

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- 629 were grasped thanks to digital view acquisition. The blade shape
- 630 change was achieved by the SMA strips embedded in the blade
- 631 and thermally activated by hot air stream flow, which reproduced
- 632 the actual automotive heat exchanger thermal ramps.

633 The comparison between the blade structures showed the influ-634 ence of the glass fiber amount on the blade stiffness. The decrease 635 in amount of glass fiber causes an increase in blade deflection dur-636 ing the activation. The blade deflection is localized at the trailing 637 edge, where the SMA strips work against the polymeric com-638 pound stiffness.

639 As a result, the SMA strips embedded in Compound A had a 640 greater influence on blade deflection and this blade structure 641 (SMA strips and Compound A) was chosen to be implemented in 642 a prototype of a heavy-duty automotive cooling axial fan.

643 The smooth blade shape variation imposed by the SMA strips 644 was compared to the step fan rotational velocity variation pro-645 vided by a viscous clutch. The results show that the SMA control 646 system can follow the thermal input (and consequently the ther-647 mal request of an automotive cooling circuit) and the time-lag is 648 less than in the case of viscous clutch. The thermal-driven SMA 649 strips can reduce the cooling circuit thermal stress as a result of a 650 smooth thermal-driven blade shape variation.

651 This preliminary analysis highlights the opportunity to generate 652 an innovative passive control system applied to an axial fan. 653 Future developments will concern the choice of the polymeric

654 compound in order to provide an increased SMA strain that leads 655 to more blade deflection.

656 In the second part of this work experimental tests and numerical 657 analyses, conducted in order to link the blade shape to the fluid 658 dynamic phenomena and the fan performance, are presented.

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- 662 The authors would like to dedicate this work in memory of
- 663 Engineer Guido Rosati, whose guidance and encouragement have
- 664 been of great importance to achieve the present results.

Nomenclature

- 665 A = austenite
- 666 m = camber
- 667 M = martensite
- 668 n = rotational velocity
- 669 t = time
- 670 T = temperature
- 671 w = heat flow

Subscripts and Superscripts 672

- 673 f = finish
- 674 s = start

675 Acronyms

- 676 H = hub
- 677 M = midspan
- PMMA = polymethyl methacrylate 678
- 679 PVC = polyvinyl chloride
- 680 S = shroud
- 681 SBTF = single blade test facility
- 682 SM = strip at midspan
- 683 SMA = shape memory alloy684
- SME = shape memory effect
- 685 SRM = stimulus-responsive material

- 686 SS = strip at shroud687 TTR = transformation temperature688
- VGC = variable geometry chevron

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