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Title: End-repair of timber beams with laterally-loaded glued-in rods:
experimental trials and failure prediction through modelling

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Abstract: Experimental trials were performed on spruce beams with one end cut and re-connected with glued-in rods to simulate repair interventions. Beams were tested in bending thus to laterally load rods, and documented by photogrammetry. Numerical analysis, adopting orthotropic constitutive laws in the context of the Continuum Damage Mechanics, was also performed. Results showed that short screws did not improve the load capacity but affected the failure mechanism. Numerical simulation correctly predicted the load-displacement curves, the damage maps, and the deflection behaviour. It was established that failure is related to a combination of perpendicular-to-grain tension, longitudinal shear and rolling shear.

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Highlights:

- Spruce beams repaired with glued-in rods were tested in bending (laterally loaded rods)
- Photogrammetric measures and numerical results obtained with an orthotropic Continuum Damage Model
- Short screws do not improve load capacity but affect the failure mechanism
- Numerical simulations predict load-displacement curves, damage maps and deflection
- Failure due to perpendicular-to-grain tension, longitudinal shear and rolling shear

30 **1 Introduction**

31 During their service life timber structures may need reinforcing interventions due to several reasons,
32 including repair when decay has occurred [1]. In fact, specific parts of timber elements (such as the ends,
33 usually in contact with masonry) are under high risk of biotic attack because excessive levels of humidity
34 may possibly develop in walls (e.g., after malfunctioning of rain-collecting systems, associated to lack of
35 adequate maintenance) [2].

36 In the events mentioned above, which can be related to both old and new buildings, several reinforcing
37 techniques can be proposed, making use of either steel or FRP elements [3–5]. Among them, the use of
38 glued-in rods (GIR) placed into predrilled holes or slots and bonded with structural adhesives, represents a
39 good solution for possible interventions; in fact, the joints prepared with GIR are considered to possess high
40 strength and stiffness along the rod axis, low cost for the preparation, improved fire resistance and good
41 aesthetic appearance, due to both hidden connecting elements and low loss of original timber [6,7]. This
42 latter characteristic is mostly important in the conservation of timber elements in heritage monuments, as in
43 this case the safeguarding authorities prescribe that the original parts are maintained integer as much as
44 possible, even when structural upgrading is needed. Thus, the use of interventions making use of GIR is a
45 realistic and successful alternative to the replacement of whole elements that are only partially decayed,
46 provided that these interventions are consistent with heritage conservation principles [8]. This is the case of
47 the Church of the Nativity in Bethlehem where this technique was recently used for the restoration of some
48 of the wooden roof trusses on the basis of indications previously provided [9].

49 However, the majority of research carried out on GIR has been oriented towards rods glued into wood
50 elements under purely axial loading. Already in these conditions bars are subjected to both uneven stress
51 distribution along their length (peaks at both rod ends) and stresses perpendicular to the axis [10–12].
52 Furthermore, in lateral loading conditions, such as those characterising the interventions on beam-ends or on
53 spliced beams (both are moment resisting connections), a multiaxial system of forces is involved, which
54 makes the stress distribution much more complex to manage. In fact, mechanical response of wood under
55 multi-axial stress states has been considered only rarely [13], and little is known about failure behaviour of
56 timber under combined stresses [14].

57 It is worthwhile observing that even in the case of uniaxial forces nominally acting on the bars the design
58 approach is not fully established among researchers, and no unique design codes exist at European level for
59 this typology [15]. Actually, Eurocode 5 [16] does not take into account this specific type of connection.
60 Thus, it can be easily understood how no reliable rules to predict the mechanical performance are yet
61 available for the case of multiaxial stress distribution. Accordingly, the usual approach encountered in the
62 scientific literature to characterise the load bearing capacity of such connections is based on developmental-
63 type investigations sometimes associated to Finite Element (FE) modelling.

64 Experimental trials were carried out on both new and old timber beams where one of the two ends were cut
65 and re-connected using GIR inserted in slots, so as to simulate a repair intervention on biologically decayed
66 ends [17]. The prepared beams were tested in bending (thus, rods were laterally loaded), and results
67 evidenced that specimens invariably failed at the repaired side and the collapse mechanism was wood
68 splitting close to rods, which was unexpected according to the adopted design approach (based on ENV
69 1995-2:1997, as in some European countries). A combined effect of shear and orthogonal-to-grain tensile
70 stresses was suggested as possible causes of collapse [17]. Splitting induced by non-axial loading was also
71 observed in other cases of laterally loaded bars. For instance, recent tests [18] were carried out on Sitka
72 spruce spliced beams, connected with glued-in basalt fibre reinforced polymer (BFRP) rods and subjected to
73 pull-bending tests (combination of axial and bending forces). Results displayed that several specimens failed
74 prematurely due to the development of splitting, and, in these cases, a lower failure strength was observed
75 with an average drop of almost 8% [18]. Splitting of GIR reinforced wood is usually attributed to: a) short
76 edge distances; b) rod misalignment; c) excessive loading perpendicular to the grain. This failure mode can
77 be prevented by transversely reinforcing the connection in the proximity of bars, e.g. by means of screws [6].
78 This solution was already adopted in tests carried out on glued-laminated spliced beams of Norway spruce
79 loaded in bending [19]. Moreover, new spruce beams with ends cut and repaired using steel bars and
80 transversal screws passing from the intrados to the extrados evidenced a 30% load bearing capacity increase
81 compared to the series without the screws [20].

82 The mechanical behaviour of such prostheses is made even more complex by the fact that in GIR-based
83 connections the joint strength depends on parameters affected by both the mechanical and the geometrical

84 factors of three different materials: wood, adhesive, and reinforcing/connecting element. Thus, FE modelling
85 is often taken into consideration to interpret the results obtained in experimental tests.

86 Actually, a reliable constitutive FE model for timber has to be anisotropic and to account for both the
87 difference in strength between tension and compression and the simultaneous presence of ductile and brittle
88 failure modes. State-of-the-art-literature on FE constitutive models mostly provides anisotropic plasticity
89 models based on Tsai and Wu [21] and Hill [22] failure criteria. For instance, Bouchaïr and Vergne [23], and
90 Clouston and Lam [24] developed single-surface plasticity models by combining the classical flow theory of
91 plasticity with the elliptical Tsai and Wu failure surface for anisotropic materials [21,25]. Other examples of
92 single-surface plasticity models are based on the failure surfaces proposed by Hoffman [26], e.g. [27].
93 Although single-surface plasticity models are able to predict the failure loads of the material, they fail to
94 reproduce different failure modes, and do not satisfactorily describe the structural behaviour. Therefore, a
95 multi-surface plasticity model has been developed [28,29]. However, the parameters employed for the
96 mathematical description of the multi-surfaces may lack of direct physical meaning, while the computational
97 transition from one surface to another one is prone to numerical instabilities. An alternative approach to
98 plasticity models consists in adopting orthotropic constitutive laws in the context of Continuum Damage
99 Mechanics (CDM). In CDM models, the progressive deterioration of the mechanical strength (due for
100 instance to the development of micro-cracks) is modelled through the introduction into the stiffness matrix of
101 damage variables representative of the state of degradation of the material, ranging from 0 for sound material
102 to 1 for completely broken material. According to CDM, when a material is unloaded after the onset of
103 damage, the stress goes to zero following a damaged elastic modulus different from the initial elastic
104 modulus of the sound material. Thus, CDM is suitable to model materials whose constitutive law exhibits a
105 brittle softening post-peak branch and failure mainly occurs by strain localization. Recently, Sandhaas et al.
106 [30] have developed a simplified effective 3D material model suitable for modelling localized ruptures
107 observed in timber structures. For a complete 3D description of wood as an orthotropic material, CDM
108 model is based on eight types of brittle and ductile failure modes governed by six damage variables [31],
109 while ductile plastic-like behaviour under compression is simulated by means of an elastic-damaging law
110 displaying a horizontal stress threshold beyond a critical strain. This approach overcomes the emergence of
111 mesh-dependency induced by the presence of softening constitutive laws through the adoption of the crack-

112 band model [32]. Alternatively, mesh dependency can be tackled by referring to non-local elastic-damaging
113 models, that introduce a regularization length by replacing a local field, such as an internal variable, with its
114 volume average or gradient expansion, e.g. [33,34]. However, besides being their implementation
115 burdensome, non-local models lead to damaged zones excessively spread over a large area as the adopted
116 regularization length is too large compared with the material characteristic length. Thus, the adoption of the
117 aforementioned crack-band model is an acceptable compromise between simplicity of implementation and
118 accuracy. Furthermore, in the Sandhaas and van de Kuilen model [31], a fictitious-viscosity-based
119 stabilization term had to be used to tackle the appearance of fast snap-back and snap-through points
120 encountered during the loading process. Although permanent plastic deformations are not captured, it has
121 been successfully applied to the modelling of timber specimens subjected to monotonic tension, compression
122 and dowel embedment [30]. Advanced elastic-plastic orthotropic constitutive laws coupled with isotropic
123 damage have been recently applied to timber elements [35,36]. However, they fail to capture the anisotropic
124 damage pattern developing in (repaired) timber beams.

125 Focus of present work is to analyse in more detail the mechanical behaviour of timber beams repaired by
126 means of solid wood prosthesis connected with steel GIR and to evaluate possible improvements of such a
127 behaviour by using screws, as suggested in [20]. In our case, differently than what was done before, the use
128 of short screws locally reinforcing the elements in the perpendicular-to-grain direction and placed close to
129 the rod insertion areas is considered in the attempt of limiting the use of long screws (mostly impacting for
130 large-sized beams). Four-point-bending tests were carried out in the Laboratory on spruce repaired beams up
131 to the final collapse, and documented by photogrammetric surveys. Such experimental tests offered the
132 opportunity of validating the mentioned 3D-CDM model to predict the load carrying capacity in bending of
133 repaired beams. In fact, to the authors' knowledge, FE simulations of timber beams end-repaired with wood
134 prosthesis, GIR steel bars and reinforcing screws have yet to be done. It is also worth evidencing that
135 adopting in present work a more robust arc-length algorithm than those generally available in commercial
136 codes made it possible to steadily follow the structural behaviour without the need of introducing spurious
137 viscosity terms of ambiguous physical meaning.

138

139

140 2 Materials and Methods

141

142 2.1 Samples and test apparatus

143 A total of 15 glulam, simply supported beams (cross section 120 x 160 mm²; span length 1800 mm), graded
144 as GL28h and made of Norway spruce (*Picea abies* Karst.) were analysed in 4-points bending tests (**Fig 1a**,
145 **1b** and **1c**). The various specimens were cut close to one end and then the two parts were re-sewed using 4
146 steel bars (S275, nominal diameter 10 mm, length 400 mm) glued into specifically prepared slots by means
147 of epoxy adhesive (Mapewood Paste 140). In this way, the variability related to wood characteristics
148 (density, slope of grain, moisture content, ageing effects etc.) on joint performances was kept to a minimum.
149 The transversal geometry of specimens is shown in **Fig 2**. Rods were placed parallel to wood grain, and slots
150 had a square section, as usual in interventions carried out onsite [37,38]. Bars were centred along each slot
151 using stoppers, thus ensuring an equal anchorage length between the two parts of the beam. Additionally, to
152 arrange a proper centring, O-rings were used at rod ends.

153 Beams were divided into three groups (each consisting of five specimens) (**Fig 1**):

154 i) (configuration 45/refer) cut inclined 45° with respect to the geometrical axis. Considering that in normal
155 practice no reinforcing screws are used, this configuration is assumed as the reference one for beams with
156 45° inclined cut (**Fig 1a**) both in experimental and numerical tests;

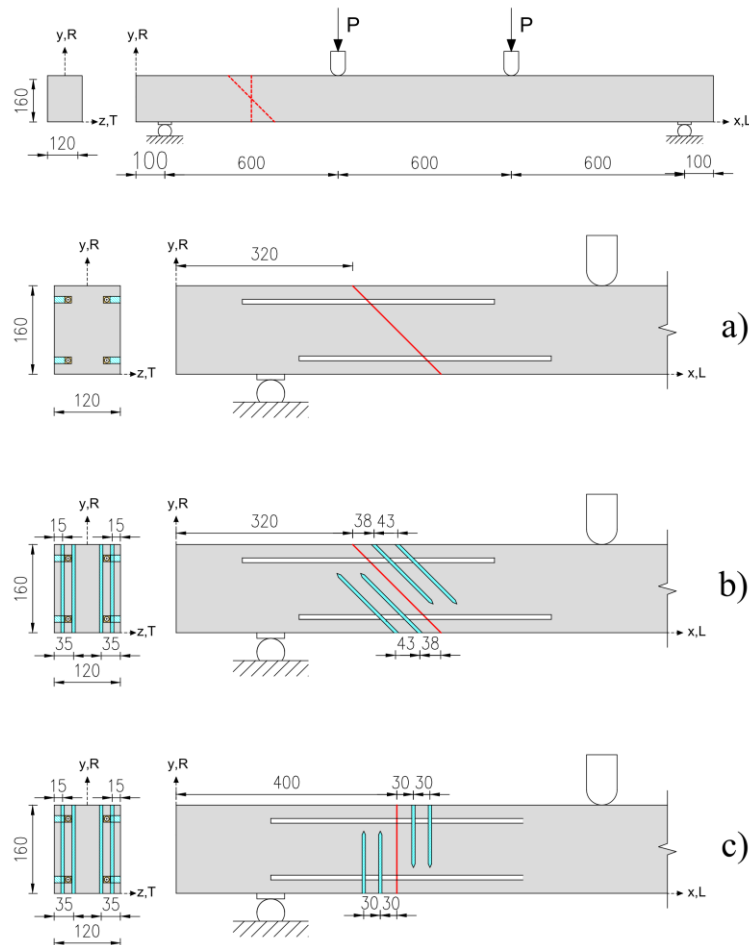
157 ii) (configuration 45/45) cut inclined 45° and use of 16 reinforcing screws (ø 6 mm, length 160 mm), 8 on
158 each part of the beam, also inclined 45° compared to the wood grain/geometrical axis. The screws insertion
159 pattern is shown in **Fig 1b** and **Fig 2**. The length of reinforcing screws (160 mm) was selected based on
160 previous tests [17], which showed that in the reference configuration the majority of specimens failed under
161 the upper bar of the main part (the longer one) of the beam and above the lower bar of prosthesis (the shorter
162 one) [17]. Therefore, the screws are stopped before the opposite rod;

163 iii) (configuration 90/90) cut inclined 90° (vertical) and use of 16 reinforcing screws (ø 6 mm, length 120
164 mm), 8 on each part of the beam, inserted perpendicular to wood grain. The screws insertion pattern is
165 shown in **Fig 1c**. The length of the screws was chosen with the same rationale as described in point (ii). It is
166 worth noting that beams with 90° inclined cut, but without screws, were not tested because previous

167 experiences proved the irrelevant influence of the cut orientation on the mechanical behaviour of the whole
168 beam [20].

169 Beams were left to cure for a minimum of 15 days in standard conditions (20°C and 65% r.h.). The 4-points
170 bending tests were configured so as to make appreciable the shear stress close to the repaired end-parts (**Fig**
171 **1**), as usually occurs in real cases (where prostheses are close to walls). Tests were carried out on a Z600
172 universal testing machine (produced by Zwick/Roell) with 600kN load capacity (load cell accuracy $\pm 1\%$).

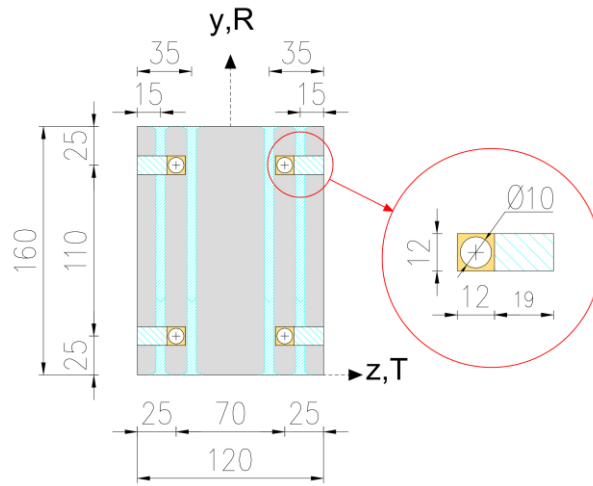
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174

175 **Fig 1** - Specimen configurations and test arrangements: (a) the reference one (cut inclined 45°, no
176 reinforcing screws); (b) same as (a) but with the use of reinforcing screws inclined 45° compared to wood
177 grain; (c) configurations with vertical cut and perpendicular-to-grain reinforcing screws.

178



179

180 **Fig 2** - Scheme of the transversal section of all repaired beams. The position of screws is also visible in the
 181 cross-section.

182

183 Firstly, beams were subjected to 2 preliminary load steps from 0 to 12 kN; then, load was gradually
 184 increased at a load-bar rate of 7.5 mm/min up to collapse. The load-displacement curve was acquired using
 185 the built-in LVDT transducer provided with the dynamometer; thus, the displacement of the upper loading
 186 anvils was actually measured. This measurement was also compared with the photogrammetric acquisitions
 187 described in Section 2.2.

188

189 *2.2 Arrangement for photogrammetric survey*

190 Many different techniques have been tested and compared over the years [39] to detect deflections, such as
 191 terrestrial laser scanners [40,41], total stations [42] and digital photogrammetry [43–45].

192 The close range photogrammetry is a no contact technique that exploits the metric data retrievable from a set
 193 of images characterized by an accuracy related to the geometric resolution and depending on the real pixel
 194 size, the so-called Ground Sample Distance (GSD). Assuming that the pixel size (ps) on the sensor plane
 195 array is known, the GSD is given by the following expression:

196
$$GSD = ps \cdot \frac{d}{f} \quad . \quad (1.1)$$

197 where d is the distance from the perspective center to the beam's surface and f is the principal distance that is
 198 very close to the focal length after the calibration process. By collecting data at a distance of about 1.5

199 meters between the camera and the surface of the beam, it is possible to reach a GSD between 0.15 and 0.35
200 mm.

201 It is also worth noting that photogrammetry allows following the deflection process up to the final collapse
202 of the beam without issues related to the range of measure due to the limited length of rod's transducers.
203 Moreover, photogrammetry allows making a global and punctual check at the same time during the whole
204 deflection process.

205 For all the above reasons, its accuracy and very high resolution at a few meter distance [46], the close range
206 photogrammetry was therefore used to perform all the deflection measurements. It is worth noting that in the
207 experimental tests presented in this paper it was necessary to take pictures of the beam at a high rate to assess
208 the cracks occurrence and evolution during the whole bending test.

209 In order to do this, 12 targets (**Fig 3**) connected to the bending machine's frame, were located on the same
210 plane of the photographed face of the beam and surveyed with a precise Leica TS06 total station. The targets
211 were assumed to be stable during the whole duration of the test.

212



213

214 **Fig 3** - Location of targets and sample beam prepared for photogrammetric detection.

215

216 To ensure that no significant temporal offset would affect the relation between deflection and actual load, a
217 well visible chronometer was placed near the left bottom targets and manually synchronized with the data
218 recorded by the load cell.

219 The photogrammetric measurement system was previously developed by simulating displacements in
220 laboratory and by detecting them through a MATLAB script. During these preliminary tests, the
221 displacements were imposed and known with an accuracy of few micrometers (**Fig 4**).

222



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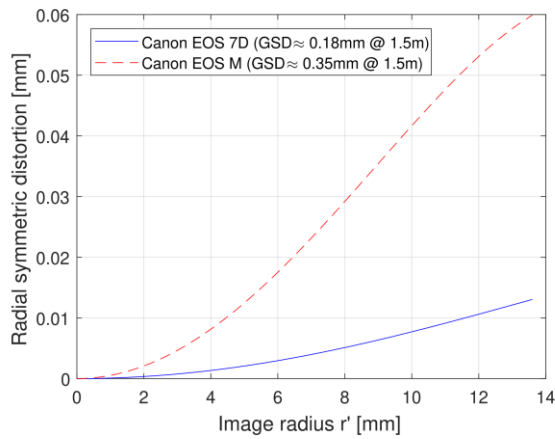
224 **Fig 4** - On the left: device for micrometric displacement imposition. On the right: image acquisition during
225 preliminary tests for automatic displacement detection.

226

227 The discrepancies between the imposed displacement and the corresponding ones turned out to be less than
228 the GSD for the whole set of measurements.

229 Two different cameras were therefore used to document each bending test: a Canon EOS 7D (18 Megapixel)
230 equipped with very low distortion lens at a fixed focal length of 35 mm for acquiring detail on the end-
231 repaired beam and a Canon EOS M with a focal length of 18 mm to capture the whole tested beam and
232 reconstruct a displacement profile. Images acquired with the first one were directly processed through the
233 MATLAB script, whereas the others were previously corrected for lens distortion and perspective by using
234 total station's surveyed targets. The reason is that both distortion and perspective deformation were proved
235 by laboratory tests to be negligible in detecting displacements in the frontal configuration with the EOS 7D
236 camera, since the specific lens used has a very low distortion (**Fig 5**) and the effect of perspective is
237 minimum. The calibration of both cameras was performed with PhotoModeler and is reported in Table 1,
238 where K_1 , K_2 , K_3 are the coefficients of Brown's model [47] and P_1 , P_2 are the decentering distortion
239 parameters.

240



241

242 **Fig 5** - Radial distortion for EOS 7D and EOS M cameras: the first one (blue) is equipped with very low
 243 distortion lens, while the second one (red) has a wide-angle lens.

244

245 A further validation of the photogrammetric acquisition system has been performed during preliminary tests
 246 carried out on selected wood beams (up to 30 kN) and using the same test arrangement: the error did not
 247 significantly exceed the GSD at any load level (data not shown).

248

249

Table 1 – Calibration parameters of the cameras.

Camera model	EOS 7D	EOS M
Principal distance [mm]	35.217	18.653
X_p (principal point x) [mm]	11.328	11.563
Y_p (principal point y) [mm]	7.672	7.399
F_w (format width) [mm]	22.681	22.740
F_h (format height) [mm]	15.113	15.164
K_1 (radial distortion 1)	8.481E-05	5.262E-04
K_2 (radial distortion 2)	-7.722E-08	-1.094E-06
K_3 (radial distortion 3)	0	0
P_1 (decentering distortion 1)	3.164E-06	-9.061E-05
P_2 (decentering distortion 2)	0	-4.208E-06
pixel size [μm]	4.375	4.387
GSD at 1.5 m distance [mm]	0.186	0.353

250

251

252 2.3 Constitutive model of the FE formulation

253 A high-performance 3D and parallelised Finite Element FORTRAN code has been developed by some of the
 254 authors to carry out large-scale numerical simulations of the experimental tests and to obtain numerical
 255 results to compare with the experimental ones. The present Fortran code adopts the orthotropic elastic-
 256 damaging constitutive model proposed in Sandhaas and van de Kuilen [31] with the introduction of the crack
 257 band model to alleviate the possible mesh-dependency of the numerical results induced by the presence of
 258 softening constitutive laws. Parallelisation made it also possible to save CPU time. The present section
 259 describes the constitutive modelling adopted in the proposed Finite Element formulation.

260 Wood is an orthotropic material and its damaging process is also orthotropic. Let the x-axis and the y-axis be
 261 aligned with the longitudinal (L) and the radial (R) directions of the wood, respectively. In the current
 262 Section, the three material directions, namely longitudinal, radial and tangential are denoted with subscripts
 263 x, y and z, respectively (**Fig 1**). Accordingly, using the Voigt notation

$$264 \quad \boldsymbol{\sigma} = [\sigma_{xx}, \sigma_{yy}, \sigma_{zz}, \sigma_{xy}, \sigma_{yz}, \sigma_{xz}]^t \equiv [\sigma_L, \sigma_R, \sigma_T, \sigma_{LR}, \sigma_{RT}, \sigma_{LT}]^t \quad (1.2)$$

265 The orthotropic non-linear behaviour of the wood has been modelled in the CDM framework according to
 266 Sandhaas et al. [30]. In particular, the six independent damage variables, one for each stress component,
 267 have been introduced and collected in the vector:

$$268 \quad \mathbf{d} = [d_L, d_R, d_T, d_{LR}, d_{RT}, d_{LT}]^t \quad (1.3)$$

269 We define the effective stress vector $\boldsymbol{\sigma}$ as the stress that acts on the effective resisting area, deprived of
 270 damaged parts, i.e.

$$271 \quad \boldsymbol{\sigma} = \mathbf{M} \boldsymbol{\sigma} \quad (1.4)$$

272 where

$$273 \quad \mathbf{M} = \begin{bmatrix} 1 & 0 & 0 & 0 & 0 & 0 \\ \frac{1}{1-d_R} & 1 & 0 & 0 & 0 & 0 \\ 0 & \frac{1}{1-d_T} & 1 & 0 & 0 & 0 \\ 0 & 0 & \frac{1}{1-d_L} & 1 & 0 & 0 \\ 0 & 0 & 0 & \frac{1}{1-d_{RT}} & 1 & 0 \\ 0 & 0 & 0 & 0 & \frac{1}{1-d_{RL}} & 1 \\ 0 & 0 & 0 & 0 & 0 & \frac{1}{1-d_{TL}} \end{bmatrix} \quad (1.5)$$

274 For the subsequent derivations, it is convenient to define the damaged compliance matrix \mathbf{C}^d of wood as

$$275 \quad \mathbf{C}^d = \begin{bmatrix} \frac{1}{(1-d_R)E_{11}} & -\frac{\nu_{21}}{E_{22}} & -\frac{\nu_{31}}{E_{33}} & 0 & 0 & 0 \\ -\frac{\nu_{12}}{E_{11}} & \frac{1}{(1-d_T)E_{22}} & -\frac{\nu_{32}}{E_{33}} & 0 & 0 & 0 \\ -\frac{\nu_{13}}{E_{11}} & -\frac{\nu_{23}}{E_{22}} & \frac{1}{(1-d_L)E_{33}} & 0 & 0 & 0 \\ 0 & 0 & 0 & \frac{1}{(1-d_{RT})G_{12}} & 0 & 0 \\ 0 & 0 & 0 & 0 & \frac{1}{(1-d_{RL})G_{13}} & 0 \\ 0 & 0 & 0 & 0 & 0 & \frac{1}{(1-d_{TL})G_{23}} \end{bmatrix} . \quad (1.6)$$

276 The nominal stress $\boldsymbol{\sigma}$ can be computed as

$$277 \quad \boldsymbol{\sigma} = \mathbf{D}^d \boldsymbol{\varepsilon} , \quad (1.7)$$

278 where the damaged stiffness matrix is obtained as $\mathbf{D}^d = (\mathbf{C}^d)^{-1}$.

279 Eight different failure criteria are considered in the adopted model, each one corresponding to a failure
280 surface in the principal stress space, as detailed in the following paragraphs.

- 281 • Criterion I - Failure in parallel-to-grain tension

282 Brittle failure mode caused by tensile stress parallel-to-grain $\tilde{\sigma}_L > 0$ and governed by the damage
283 variable d_L . It obeys the loading-unloading inequality

$$284 \quad F_I(\boldsymbol{\sigma}) = \frac{\tilde{\sigma}_L}{f_{t,L}} - 1 \leq 0 , \quad (1.8)$$

285 where $f_{t,L}$ is the tensile strength along the longitudinal direction.

- 286 • Criterion II - Failure in parallel-to-grain compression

287 Ductile failure mode caused by compressive stress parallel-to-grain $\tilde{\sigma}_L < 0$ that affects the damage
288 variable d_L . The failure is associated with the loading-unloading inequality

$$289 \quad F_{II}(\boldsymbol{\sigma}) = \frac{-\tilde{\sigma}_L}{f_{c,L}} - 1 \leq 0 \quad (1.9)$$

290 where $f_{c,L}$ is the compressive strength along the longitudinal direction.

- 291 • Criterion III/IV - Failure in perpendicular-to-grain tension

292 Brittle failure modes governed by tensile perpendicular-to-grain ($\tilde{\sigma}_R > 0$ and $\tilde{\sigma}_T > 0$), longitudinal
293 shear ($\tilde{\sigma}_{LR}, \tilde{\sigma}_{LT}$) and rolling shear ($\tilde{\sigma}_{RT}$) stresses. They are characterized by the development of
294 splitting cracks in LR-plane and LT-plane, respectively, and are related to the damage variables d_R ,
295 d_T , d_{LR} , d_{LT} and d_{RT} . The quadratic loading-unloading inequalities:

$$296 F_{III}(\sigma) = \frac{\tilde{\sigma}_R^2}{f_{t,R}^2} + \frac{\tilde{\sigma}_{LR}^2}{f_{v,LR}^2} + \frac{\tilde{\sigma}_{RT}^2}{f_{v,RT}^2} - 1 \leq 0 \quad \text{and} \quad F_{IV}(\sigma) = \frac{\tilde{\sigma}_T^2}{f_{t,T}^2} + \frac{\tilde{\sigma}_{LT}^2}{f_{v,LT}^2} + \frac{\tilde{\sigma}_{RT}^2}{f_{v,RT}^2} - 1 \leq 0 \quad , \quad (1.10)$$

297 are adopted, where $f_{t,R}$ is the tensile strength along the radial direction, $f_{t,T}$ is the tensile strength
298 along the tangential direction, $f_{v,LR}$ is the shear strength along the LR plane, $f_{v,LT}$ is the shear
299 strength along the LT plane, and, finally, $f_{v,RT}$ is the rolling shear strength of the material.

- 300 • Criterion V/VI - Failure in perpendicular-to-grain compression

301 Ductile failure modes caused by the compressive stress along the radial or the tangential direction
302 $\tilde{\sigma}_R < 0$ and $\tilde{\sigma}_T < 0$, and governed by the damage variables d_R and d_T . They are associated with
303 the following loading-unloading inequalities:

$$304 F_V(\sigma) = \frac{-\tilde{\sigma}_R}{f_{c,R}} - 1 \leq 0 \quad \text{and} \quad F_{VI}(\sigma) = \frac{-\tilde{\sigma}_T}{f_{c,T}} - 1 \leq 0 \quad , \quad (1.11)$$

305 where $f_{c,R}$ and $f_{c,T}$ are the compressive strength along the radial direction and along the tangential
306 direction, respectively.

- 307 • Criterion VII/VIII - Shear failure

308 Brittle failure modes induced by the longitudinal shear stress $\tilde{\sigma}_{LR}$ and $\tilde{\sigma}_{LT}$ and the rolling shear
309 stress $\tilde{\sigma}_{RT}$, in the case that a compressive perpendicular-to-grain $\sigma_{RT} < 0$ occurs. They are
310 governed by the damage variables d_{LR} , d_{LT} and d_{RT} . The current case is governed by the quadratic
311 loading-unloading inequalities:

$$312 \quad F_{VII}(\sigma) = \frac{\tilde{\sigma}_{LT}^2}{f_{v,LT}^2} + \frac{\tilde{\sigma}_{RT}^2}{f_{v,RT}^2} - 1 \leq 0 \quad \text{and} \quad F_{VIII}(\sigma) = \frac{\tilde{\sigma}_{LT}^2}{f_{v,LT}^2} + \frac{\tilde{\sigma}_{RT}^2}{f_{v,RT}^2} - 1 \leq 0 \quad . \quad (1.12)$$

313 In general, the loading-unloading Kuhn-Tucker conditions for the multi-surface CDM model are cast as

$$314 \quad \phi_m(\boldsymbol{\sigma}, k_m) = F_m - k_m \leq 0 \quad \dot{k}_m \geq 0 \quad \phi_m \dot{k}_m = 0 \quad m = I, \dots, VIII \quad , \quad (1.13)$$

315 where k_m is the state variable related to the m -th failure criteria, and \dot{k}_m denotes the increment of
316 k_m evaluated with respect to a fictitious time parameter.

317 In order to simulate both brittle and ductile failure modes, the Sandhaas et al. model [30] considers two
318 damaging laws. In particular, Fig 6a shows a qualitative representation of the adopted stress-strain law
319 adopted for each case of ductile failure mode, such as failure by compression in the parallel-to-grain and
320 perpendicular-to-grain directions. In this case, the relevant damage variable, that is activated when the
321 loading function is greater than zero, is calculated as:

$$322 \quad d_i(k_m) = 1 - \frac{1}{k_m} \quad i = L, R \quad . \quad (1.14)$$

323 For each brittle failure mode induced by tensile and/or shear stresses, the bilinear softening stress-strain law
324 displayed in **Fig 6b** is used. In this case, the damage variables related to tensile failure modes are computed
325 according to:

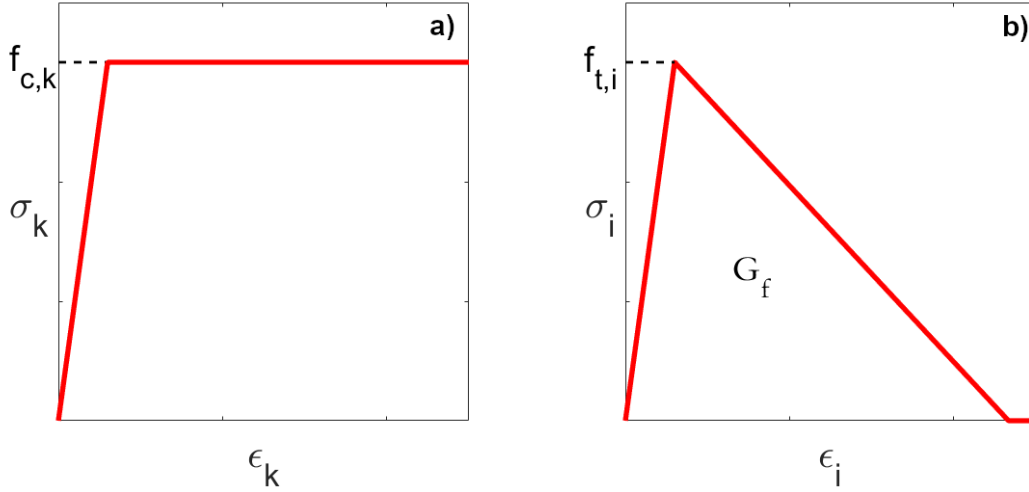
$$326 \quad d_i(k_m) = 1 - \frac{1}{f_{t,i} - 2g_{f,i}E_i} \left(f_{t,i} - \frac{2g_{f,i}E_i}{k_m} \right) \quad i = L, R, T \quad , \quad (1.15)$$

327 while, for shear failure modes, the damage variables are defined as:

$$328 \quad d_i(k_m) = 1 - \frac{1}{f_{v,i} - 2g_{f,v,i}G_i} \left(f_{v,i} - \frac{2g_{f,v,i}G_i}{k_m} \right) \quad i = LR, LT, RT \quad , \quad (1.16)$$

329 $g_{f,i}$ and $g_{f,v,i}$ being the specific fracture energies for the tensile and the shear failure modes, respectively.

330



331

332 **Fig 6** – Qualitative representation of the stress-strain laws adopted for: a) ductile failure modes, where $k =$

333

L, R, T ; b) brittle failure modes, where $i = L, R, T, LR, LT, RT$.

334

335 Mesh-dependency is alleviated by adopting the crack band model [32], where the specific fracture energy

336 g_f of the single FE e is related to the fracture energy of the material G_f and to the characteristic size of

337

the finite element h_e through

338

$$g_f = \frac{G_f}{h_e} \quad (1.17)$$

339

Fracture energies have been taken from Sandhaas and van de Kuilen [31] and Mackenzie-Helnwein et al.

340

[28] and the other properties from Kretschmann [48] (Table 2).

341

342

Table 2 – Material parameter of Norway spruce employed for the numerical simulations.

Elastic Properties [MPa]				
E_L	E_T	E_R	$G_{LT} = G_{LR}$	G_{RT}
11000	500	800	690	50
Strengths [MPa]				
$f_{t,L}$	$f_{c,L}$	$f_{t,R} = f_{t,T}$	$f_{c,L} = f_{c,T}$	$f_{v,RT} = f_{v,RL} = f_{v,RL}$
45 - 65	25 - 35	0.5 - 2.5	3.8	4 - 6.9
Fracture Energies [N/mm]				
$G_{f,L}$	$G_{f,R} = G_{f,T}$	$G_{f,v,RL}$	$G_{f,RT}$	
2	0.3	1.2	0.6	

343

344 The 3D FORTRAN FE code was validated by modelling experimental 4-point bending tests carried out on
345 four integer Norway spruce beams. The experimental and numerical structural responses were in close
346 agreement. In particular, not only the load vs displacement curves were highly comparable (general shape,
347 peak loads and displacements at failure correctly predicted), but also the failure mode associated with
348 bending failure was consistently reproduced: indentation under the anvils, damage slowly growing at
349 extrados due to compressive stress, damage due tension quickly propagating at the intrados under the loaded
350 zone in the post-peak regime (data not shown).

351

352

353 **3 Results and discussion**

354 *3.1 Experimental results and photogrammetric measurements for end-repaired beams*

355 The results of the tests carried out in the Laboratory are shown in Table 3. The experimental data were
356 compared to the expected mean values for timber of class GL28h, estimated by considering the product of
357 the characteristic values by 1.3, which is normally regarded as representing the mean value of a given timber
358 property. Average experimental loads at collapse, i.e., the maximum loads reached during each test (Table
359 3), evidenced how these values were slightly lower, although comparable, with the mean value expected on
360 the basis of calculations (62.12 kN). This occurrence also reflected on the average flexural stress at collapse,
361 calculated considering a continuous (i.e., not cut) wood section, according to the usual expressions for 4-
362 points bending tests. The experimental average values were included in the range 34.7-36.1 MPa, slightly
363 lower than the expected mean value for timber (36.4 MPa). In fact, only those specimens characterised by
364 the presence of timber-related defects (knots, localised grain deviation, rupture in finger joints etc., observed
365 in series 45/45 and 90/90, Table 3) broke close to the central part of beams, i.e., in the nominally pure
366 bending area.

367 Moreover, considering that our experimental test arrangement was configured so as to make appreciable the
368 shear stress close to the repaired end-parts (**Fig 1**), it is worth evidencing that the average values of shear
369 stress at collapse, also calculated considering a continuous wood section, that is using the expression

370 $\tau_R = 1.5 \cdot \frac{F_{\max}}{2 \cdot b \cdot h}$, were included in the range 2.3-2.4 MPa, that is below the characteristic value for the

371 considered class (3.5 MPa).

372 In general, it is possible to state that, although the expected values for the integer beams could not be fully
 373 recovered using the adopted technique, they were very close to them.

374 Concerning the comparison among the considered intervention typologies, some statistical evaluations were
 375 carried out using Student-t test ($\alpha=0.05$) to take account of the usual variability affecting tests on timber.
 376 This test showed that the differences in average values of both loads at collapse and stiffness (proportional to
 377 the slope of load-deflection curves assessed at mid-span, Table 3) were not statistically significant
 378 ($p \gg 0.05$). This finding clearly indicated that the presence of short screws did not improve the mechanical
 379 performance of repaired elements, but they left unaltered both the load at collapse and the stiffness of
 380 reinforced beams. Looking at the failure mode, it was observed that beams of series 45/refer collapsed due to
 381 the occurrence of splitting invariably localised under the upper rods of the main (longer) part of the beam,
 382

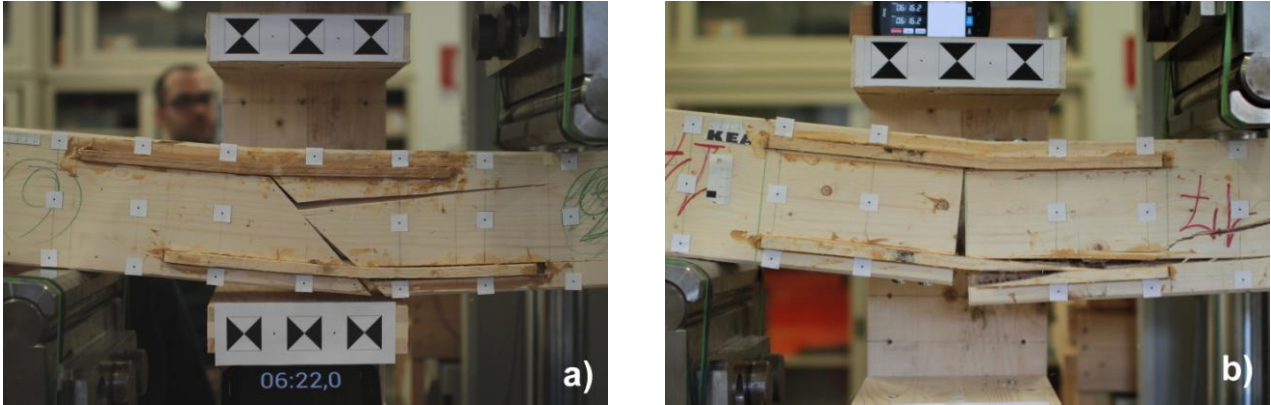
383 Table 3 - Experimental results obtained for end-repaired beams. UUBB = under the upper bar of the main
 384 (longer) part of the beam; ALBB = above the lower bar of the main part of the beam; POLB = pull-out of the
 385 lower bars; T = timber-related defects (knots, localised grain deviation, rupture in finger joints etc.). In bold
 386 the mechanism observed in the pre-peak regime.

Configuration	Sample	Pmax [kN]	Deflection @ Pmax [mm]	Slope of the load-deflection curve @ mid span, N/mm	Failure modes
45 / refer	T6	65.64	19.7	4244	UUBB ; POLB
	T9	59.01	18.1	3802	UUBB ; POLB
	T10	65.08	19.7	3874	UUBB ; POLB
	T12	57.00	19.8	3583	UUBB ; POLB
	Average	61.68	19.3	3876	
45 / 45	T1	51.26	14.7	4004	T
	T5	63.36	17.7	4129	ALBB ; POLB
	T8	49.87	15.4	3557	T
	T11	65.30	18.1	4302	ALBB ; POLB
	T13	66.67	N.A.	3730	ALBB ; POLB
Average	59.29	16.5	3945		
90 / 90	T2	55.94	23.1	3452	T
	T15	62.35	28.5	3385	ALBB
	T16	56.23	17.5	4191	ALBB
	T17	64.46	21.4	4241	ALBB
	T18	58.89	18.2	4086	ALBB
Average	59.57	21.7	3871		

387

388 similar to Pizzo et al. [17]. This mechanism was also associated to the pull-out of lower bars, which occurred
 389 at the moment of collapse. Instead, in both 45/45 and 90/90 series cracks due to wood splitting moved down
 390 towards lower bars, localised at the tips of screws (mechanism ALBB in Table 3). Thus, the presence of
 391 screws (although short) affected the failure mechanism of repaired beams.

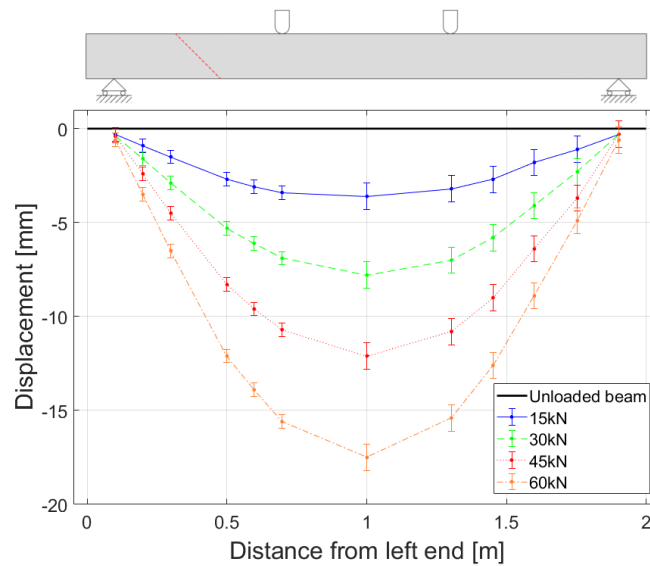
392



393

394 **Fig 7** – Examples of failure modes for representative elements: a) UUBB; b) ALBB.

395



396

397 **Fig 8** - Photogrammetrically detected deflections of one representative beam of series 45/refer (error bars
 398 have been magnified by a factor of two). The left end measurements are more precise due to the smaller
 399 pixel size on Canon EOS 7D pictures.

400

401 Through the photogrammetric workflow shown in Section 2.2, the displacement of every single marker dot
 402 on the beam could be accurately assessed. In particular, the displacement of the intrados points was detected
 403 to reconstruct the deformation profile of the beam (**Fig 8**) at a certain applied load step in order to allow
 404 some comparisons with the corresponding values computed by the numerical model. The load step increment
 405 adopted for this purpose is 15kN.

406

407 *3.2 Numerical results for the end-repaired beams*

408 The three experimental typologies shown in **Fig 1** have been simulated, and the main mechanical parameters
 409 obtained after numerical analysis shown in Table 4. The steel rods and the adhesive have been modelled as
 410 elastic-plastic isotropic materials with the material parameters provided by the manufacturers. In particular,

411

412 Table 4 - Peak loads, displacements at peaks and failure modes for each numerically simulated
 413 configuration.

Configuration	Peak Load [kN]	Displacement [mm]	Failure Mode
45/refer	62.02	17.10	Pull-out of the lower bar in the main beam combined with horizontal crack above the lower bars in the main part of the beam
45/45	61.72	16.46	Pull-out of the lower bar in the main beam combined with horizontal crack above the lower bars in the main part of the beam
90/refer	52.78	15.03	Yielding of the lower bars
90/90	56.43	18.14	Horizontal crack above the lower bars in the main beam
90/90 passing screws S275	60.02	20.38	Yielding of the lower bars
90/90 passing screws B450C	66.64	22.79	Pull-out of the lower bar in the main beam combined with horizontal crack above the lower bars in the main beam
45/refer $f_{t,R} = 0.50$ MPa $f_v = 6.9$ MPa	57.16	17.70	Shear cracks across the cut and horizontal crack above the lower bars in the main beam
45/refer $f_{t,R} = 1.25$ MPa $f_v = 6.9$ MPa	62.86	18.19	Pull-out at the end of the lower bar in the main beam
45/refer $f_{t,R} = 2.50$ MPa $f_v = 6.9$ MPa	63.87	17.77	Pull-out at the end of the lower bar in the main beam

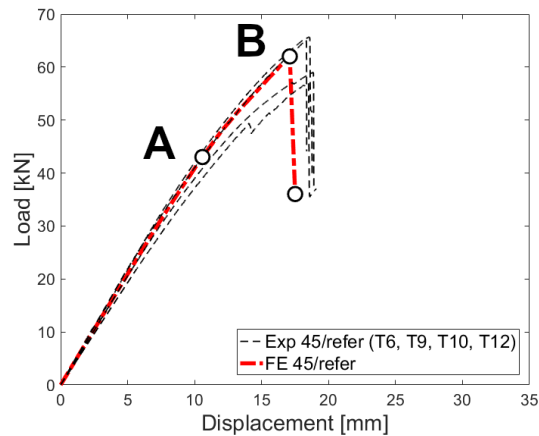
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415

416 the steel and adhesive Young's moduli are 210000 and 4000 MPa, respectively, while the Poisson
 417 coefficients are 0.3 and 0.35, respectively. Moreover, steel S275 has been used in all the simulations, except
 418 in the case of passing screws as illustrated in Section 3.2.2, where both steel S275 and B450C have been
 419 used. For the sake of simplicity, a perfect adhesion has been assumed at the steel-adhesive and the adhesive-
 420 wood interfaces. A thin layer made of a linear elastic isotropic material with reduced Young's modulus has
 421 been introduced to simulate the unilateral contact at the cut-interface.

422 As for wood, the parallel-to-grain compressive strength, $f_{c,L}$, and the tensile strength, $f_{t,L}$, have been set to
 423 35 and 65 MPa, respectively, whereas the other parameters are shown in Table 2. More comments on the
 424 model sensitivity to the material parameters are given in Section 3.2.4.

425



426

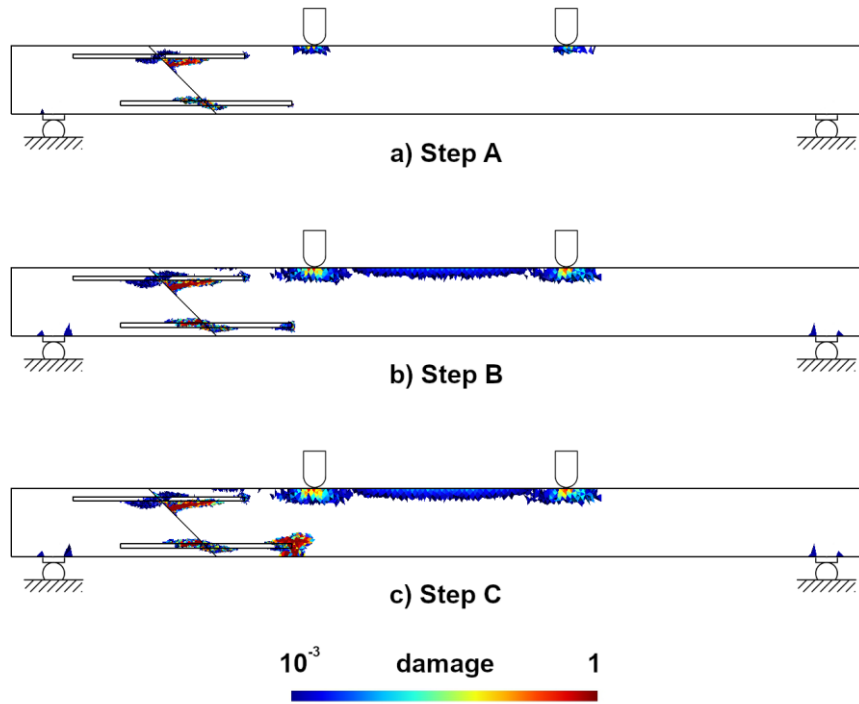
427 **Fig 9** - Comparison between experimental (black dashed line) and numerical (red point-dashed line) load vs
 428 deflection curves obtained for 45/refer beam and obtained with $f_{t,L} = 35$ MPa, $f_{c,L} = 65$ MPa, $f_{t,R} = 1.25$
 429 MPa, $f_v = 4$ MPa.

430

431 A three-dimensional mesh with 228599 tetrahedrons and a characteristic mesh size of 2 mm has been used.
 432 The numerical test took 13 hours. Owing to the symmetry with respect to the LR plane, only one half of the
 433 specimen has been modelled.

434 The experimental and the numerical load vs displacement curves for the 45/refer configuration of **Fig 1a** are
 435 displayed in **Fig 9**. It can be drawn that the response of the repaired beam is brittle.

436 **Fig 10** displays the contour plots of the cumulative damage greater than 10^{-3} evaluated at the development of
437 the first macro-crack (Step A), at the peak load (Step B) and at complete failure (Step C).
438



439
440 **Fig 10** - Contour plots of the cumulative damage for the 45/refer beam (**Fig 1a**) at different steps of the load
441 vs displacement curve (**Fig 9**).
442

443 Before the load peak, the contour map of the cumulative damage at Step A displays two horizontal cracks
444 located below the upper bar in the main part of the beam and above the bottom bar in the prosthesis.
445 Furthermore, an indentation process develops under the loaded area. Then, at step B corresponding to the
446 peak reached at load 62.02 kN, the compression generated by the bending starts damaging the upper side of
447 the beam section, while the aforementioned horizontal cracks further propagate. In addition, a damaged zone
448 close to the left end of the lower bar is activated. This latter damage zone spreads at failure, reached at step
449 C, possibly, as a consequence of the pull out of the bars.

450 **Fig 9** and **Fig 10** display a good agreement between the numerical and the experimental results. Both the
451 failure mode and the structural response obtained from the numerical simulation are consistent with those
452 observed in the experimental tests. Particularly interesting is the agreement with the cumulative damage at

453 Step A, where the cracks developing before the maximum load is attained (evidenced both in present and
 454 previously reported tests, [17]) are correctly predicted.

455

456 3.2.1 Comparison between numerical results and photogrammetric measurements

457 The displacements of selected points close to the intrados were detected to reconstruct the deformation
 458 profile of the 45/refer beam at a predefined load step of 15 kN (Table 5). These displacements are then
 459 compared with the values computed by the numerical model. The experimental and the numerical
 460 displacement data are in good agreement.

461

462 Table 5 - Comparison between the numerical and the experimental displacement profile for different values
 463 of the applied loads for the 45/refer series.

	P=15kN		P=30kN		P=45kN		P=59 kN	
<i>x</i> [mm]	PHOT [mm]	FEM [mm]	PHOT [mm]	FEM [mm]	PHOT [mm]	FEM [mm]	PHOT [mm]	FEM [mm]
100	0.30 ± 0.18	0.1422	0.40 ± 0.18	0.2872	0.40 ± 0.18	0.5161	0.60 ± 0.18	0.9764
199	0.90 ± 0.18	0.7262	1.60 ± 0.18	1.4668	2.40 ± 0.18	2.5740	3.50 ± 0.18	4.7429
299	1.50 ± 0.18	1.3408	2.90 ± 0.18	2.7084	4.50 ± 0.18	4.7860	6.00 ± 0.18	8.8832
500	2.70 ± 0.18	2.6529	5.30 ± 0.18	5.3595	8.30 ± 0.18	9.5353	12.10 ± 0.18	17.9188
597	3.10 ± 0.18	3.0720	6.10 ± 0.18	6.2056	9.60 ± 0.18	11.3841	13.90 ± 0.18	21.7272
697	3.40 ± 0.35	3.4578	6.90 ± 0.35	6.9847	10.70 ± 0.35	12.8380	15.60 ± 0.35	24.6683
1001	3.60 ± 0.35	3.9006	7.80 ± 0.35	7.8787	12.10 ± 0.35	13.2207	17.50 ± 0.35	23.5521
1303	3.20 ± 0.35	3.4173	7.00 ± 0.35	6.9023	10.80 ± 0.35	11.2025	15.40 ± 0.35	19.2159
1452	2.70 ± 0.35	2.7500	5.80 ± 0.35	5.5543	9.00 ± 0.35	8.9089	12.60 ± 0.35	15.0855
1600	1.80 ± 0.35	1.9824	4.10 ± 0.35	4.0041	6.40 ± 0.35	6.3709	8.90 ± 0.35	10.6981
1754	1.10 ± 0.35	1.0235	2.30 ± 0.35	2.0673	3.70 ± 0.35	3.2646	4.19 ± 0.35	5.4380

464

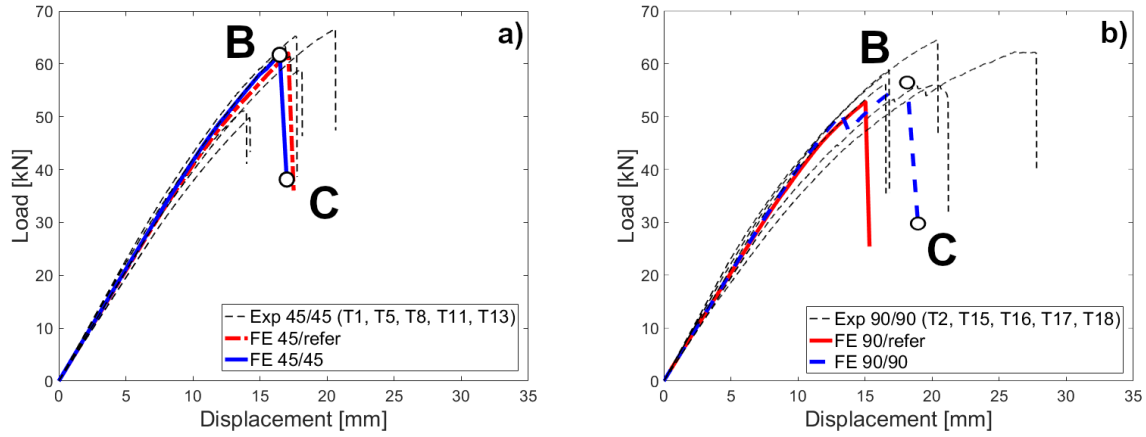
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466 3.2.2 Influence of the screws

467 The simulation of the two numerical models related to configurations 45/45 and 90/90 (respectively **Fig 1b**
 468 and **Fig 1c**) was made by 315484 and by 297092 tetrahedrons, respectively. It has taken almost 19 hours for
 469 the 45/45 configuration and 17 hours for the 90/90 one on a PC i7-5930k CPU.

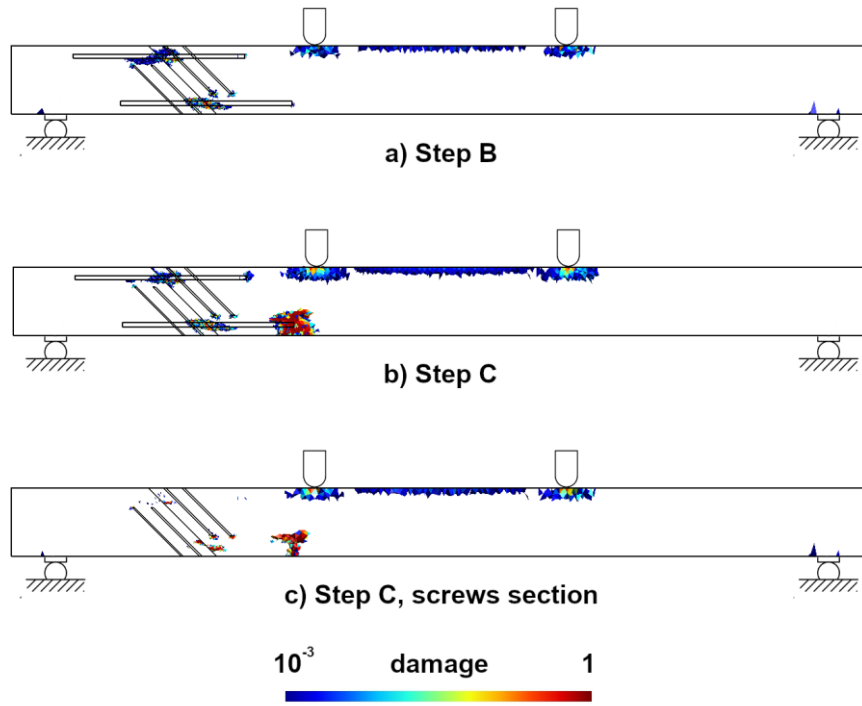
470 The numerical and the experimental load vs deflection curves are reported in **Fig 11**. For the sake of
 471 completeness, the results obtained for the configuration without reinforcing screws have also been reported.

472 It can be deduced that the screws do not significantly affect the peak load and the brittleness of the structural
 473 response of the 45/refer end-repaired beams, whereas they increase the peak load by only 6% in the 90/90
 474 configuration.
 475



476
 477 **Fig 11** - Comparison between experimental (black dashed line) and numerical load vs displacement curves
 478 for: a) the 45/refer (red point-dashed line) and the 45/45 (blue continuous line); b) the 90/refer (red continuous
 479 line) and the 90/90 (blue dashed line) configurations. $f_{t,L} = 35$ MPa, $f_{c,L} = 65$ MPa, $f_{t,R} = 1.25$ MPa, $f_v = 4$
 480 MPa.

481
 482 **Fig 12** and **Fig 13** show the contour plots of the cumulative damage for the 45/45 and 90/90 configurations,
 483 respectively. In particular, the displayed contour plots correspond to the stages immediately before (a) and at
 484 failure (b), namely at steps B and C (**Fig 11**) according to the notation adopted for the 45/refer configuration.
 485 Furthermore, a detail of the cumulative damage at failure in the screws section is reported in **Fig 12c** and **Fig**
 486 **13c**. These contour plots show that, in both the considered configurations, the reinforcing screws stop the
 487 crack propagation below the upper rods of the main part of the beam, whereas some damage develops at the
 488 screws tips. Moreover, failure occurs around steel bars, as a consequence of either yielding or pull-out-
 489 related tensions.
 490

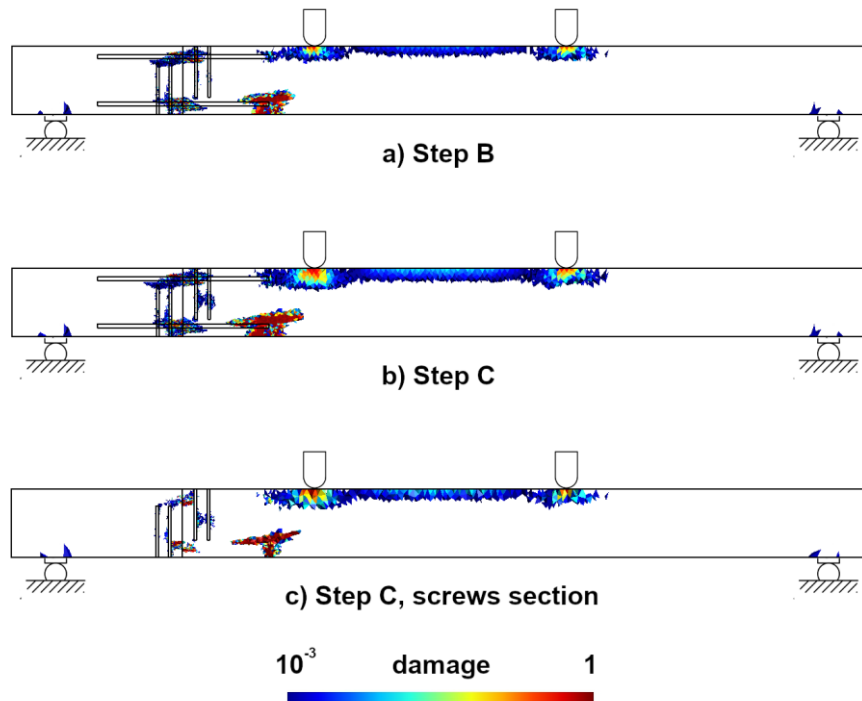


491

492 **Fig 12** – Contour plots of the cumulative damage immediately before (a) and at failure (b) for the 45/45

493 configuration (**Fig 1b**). In (c), the cumulative damage at failure in the screws section is shown.

494



495

496 **Fig 13** - Contour plots of the cumulative damage immediately before (a) and at failure (b) for the 90/90

497 configuration of **Fig 1c**. In (c), the cumulative damage at failure in the screws section is shown.

498

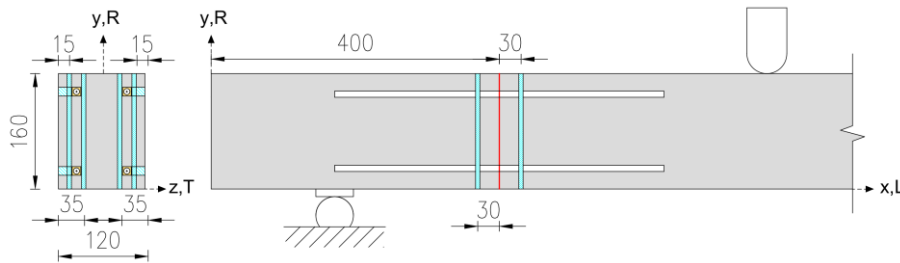
499 The results of numerical simulations confirm the experimental evidence that the presence of short reinforcing
500 screws on the load vs displacement curves is marginal for both the 45/45 and 90/90 series. The numerical
501 damage map is also in agreement with the experimentally observed cracking pattern, displaying the
502 movement of the damage zone at failure from the upper bars of the main part of the beams towards the lower
503 ones.

504

505 3.2.3 Numerical assessment of the effect of passing screws

506 Numerical investigation was also used to assess to which extent the usage of passing screws (**Fig 14**) might
507 further improve the structural performance of the end-repaired beams. For this purpose, simulations have
508 been carried out for the case of the 90° cut. In detail, two simulations have been run, one with a steel with
509 yielding stress equal to 275 MPa, the same as used on the experimental tests, and another one with a steel
510 yielding at 450 MPa, the same used in [20], with the aim of assessing the influence of the steel strength on
511 the increase of the peak load. The failure modes detected in the simulations are presented in Table 4.

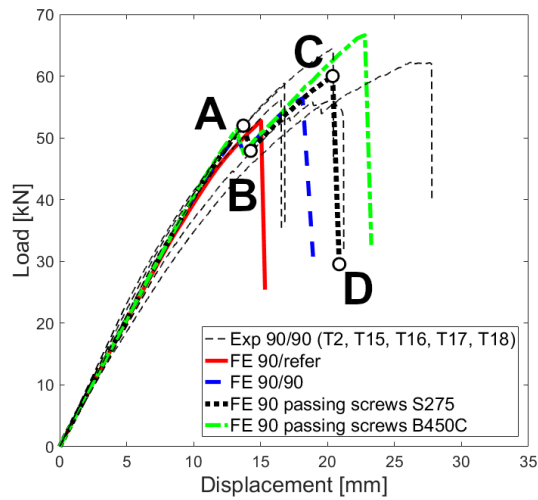
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513

514 **Fig 14** – The configuration with 90° cut and passing screws simulated in the numerical investigation.

515



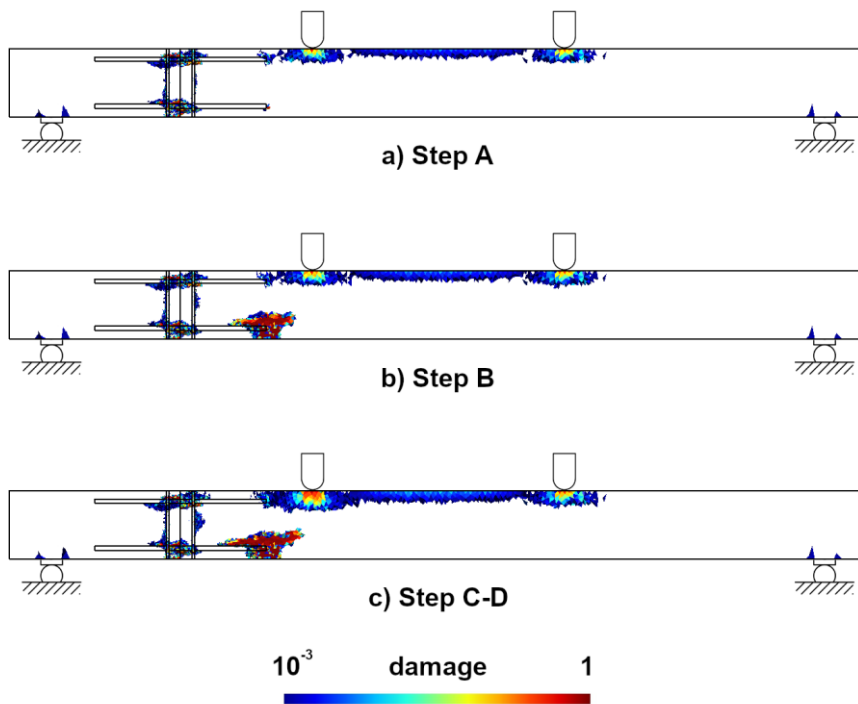
516

517 **Fig 15** – Comparison between numerical load vs displacement curves for the 90° cut beam with passing

518 screws in both cases of steel S275 and B450C, the 90/90 configuration with non-passing screws

519 (experimental tests), and the 90/refer beam. $f_{t,L} = 35$ MPa, $f_{c,L} = 65$ MPa, $f_{t,R} = 1.25$ MPa, $f_v = 4$ MPa.

520



521

522 **Fig 16** - Contour plots of the cumulative damage for the 90° cut beam with passing screws and steel S275.

523

524 **Fig 15** shows that, with respect to the 90/refer beam, the passing screws increase the failure load by 14% in

525 the case of steel S275, namely from 52.78 kN to 60.02 kN, and by 26% in the case of steel B450c, that is

526 from 52.78 kN to 66.64 kN. The damage maps illustrated in **Fig 16** display that, analogously to the non-
527 passing screws, the presence of the passing screws inhibits damage in their surrounding zone: failure starts at
528 step C, completes at step D and occurs by yielding and/or pull-out of the steel bars. On the other hand, it is
529 worth noting that, when screws made of B450C are simulated, the damaged zone developing above the lower
530 bar in the main part of the beam is larger, and failure mainly occurs by pull-out of the lower bar, while the
531 yielding of the latter is still present but to a lesser extent.

532

533 3.2.4 Numerical sensitivity analysis to the orthogonal-to-the-grain tensile strength

534 A sensitivity analysis has been carried out on the 45/refer configuration to assess the influence of the
535 strength properties along the direction orthogonal to the grain, having fixed all the remaining material
536 parameters. In particular, the cases where $f_{t,R} = 0.50, 1.25, 2.50$ MPa with $f_v = 6.9$ MPa have been
537 considered. **Fig 17** compares the experimental and the structural paths computed with the aforementioned
538 parameters. It can be drawn that changing the strength in the radial direction does not significantly alter the
539 load-displacement curves. The corresponding damage maps at failure are displayed in **Fig 18**, and they
540 evidence that the respective damage patterns are markedly different, especially in the case of $f_{t,R} = 0.50$
541 MPa, for which the model predicts diffused shear cracks across the cut that, however, have not been
542 experimentally observed. This circumstance implies that the characteristic value of 0.5 MPa recommended in
543 codes for class GL28h (the same used in present tests) is highly precautionary and not suitable for numerical
544 codes. The failure modes computed for the whole set of constitutive parameters are collected in Table 4. A
545 satisfying correspondence between the numerical and the experimental failure modes can be evidenced.

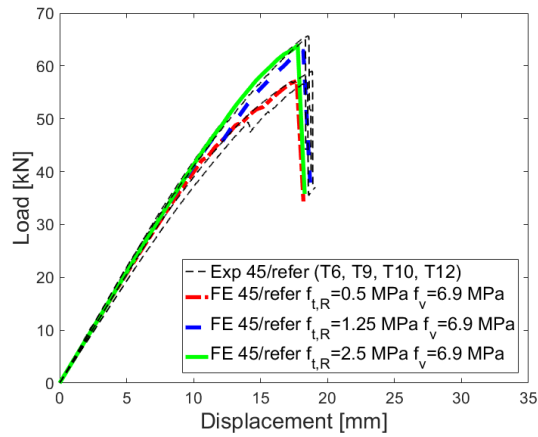
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547

548 **4 Conclusions**

549 Some experimental tests were aimed at evaluating the effectiveness of short screws inserted in beams
550 repaired at one end with the use of GIR, to locally reinforce timber in the perpendicular-to-grain direction.
551 Moreover, a comprehensive numerical analysis, supported by photogrammetric measurements, was carried
552 out in order to interpret the experimental behaviour of both reinforced and unreinforced beams.

553



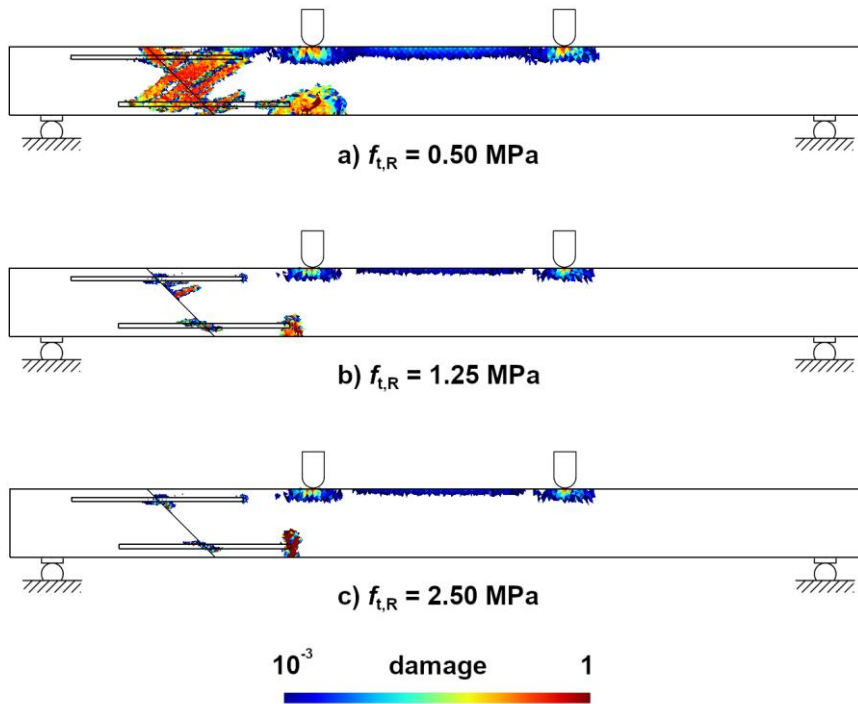
554

555 **Fig 17** - Sensitivity analysis: Influence of the radial tensile strength. $f_{t,R}$ obtained with $f_{t,L} = 35$ MPa,

556

$$f_{c,L} = 65 \text{ MPa.}$$

557



558

559 **Fig 18** - Contour plots of the cumulative damage at failure for the 45/refer beam for variable $f_{t,R}$.

560

561 Results showed that short screws were not able to substantially improve the load capacity of repaired
 562 elements, for both configurations 45/45 and 90/90. However, the presence of screws (although short)
 563 affected the failure mechanism of repaired beams: the movement of the due-to-splitting cracks towards the

564 lower bars (of the main part of the beam) was observed in the two screw-reinforced configurations,
565 differently from the 45/refer series where splitting was localised below the upper bars.

566 Numerical simulation agreed to a very good extent with experimental results (for both screw-reinforced and
567 unreinforced configurations): both the load-displacement curves and the obtained damage maps observed in
568 mechanical tests were correctly predicted, and the deflection behaviour was consistent with the one
569 registered in the photogrammetric survey. Moreover, even when applied to the case, already reported in the
570 literature, of screws passing from extrados to intrados, the numerical model predicted a load capacity
571 increase very close to the one previously reported. Finally, sensitivity analysis allowed establishing that the
572 experimentally observed failure mechanism is related to the failure criteria III/IV, which in fact provides a
573 combination of tension perpendicular-to-grain, longitudinal shear and rolling shear stresses.

574

575

576 **Acknowledgements**

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579 Mechanical Tests at CNR-IVALSA for their helpful assistance during tests.

580

581

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583

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